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TECHNICAL REPORT ARLCD-CR-83045

M577 MODULAR SETBACK PIN COMPUTER SIMULATION AND AUTOMATED ASSEMBLY AND TESTING

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MARCH 1984

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DOVER, NEW JERSEY

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M577 fuze

Modular setback pin

MSP computer simulation

MSP automated assembly and testing equipment

Die cast MSP

20. ABSTRACT (Continue on reverse side if necessary and identity by block number)

The objective of this project was to conduct a computer simulation of the modular setback pin (MSP) motion under setback. The simulation relates time to function, setback magnitude, and nine MSP design features. A second objective was to evaluate possible methods of automating the assembly and testing of the MSP. A history of previous ballistic testing and design changes is also included.

CONTENTS

		Page
TION		1
DN .		5
SIMULATION		7
D ASSEMBLY		10
URABILITY		14
ING TEST PROGRAM		15
ONS AND RECOMMENDATIONS		23
ES		
A. REQUIRED DRAWING CHANGES		
B. MODULAR SETBACK PIN EQUATIONS OF MOTION		
C. MODULAR SETBACK PIN COMPUTER SIMULATION PROGRAM		
D. EVALUATION OF COMPUTER SIMULATION PARAMETERS		
. SETBACK MAGNITUDE VERSUS TIME REQUIRED FOR MSP TO FUNCTION		
A RELATIVE COMPARISON OF NINE DES RELATE TO DROP HEIGHT	IGN FEATURES AS THEY	63
ETCHING PROCEDURE FOR THE MSP HOU	SING	91
M577 MODULAR SETBACK PIN BALLISTI	C TESTING SUMMARY	94
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FIGURES

		rage
1	Modular Setback Pin Exploded View	2
2	#4 Plate Mating Geometry	3
3	Assembly Procedure	4
4	Lock Down Feature of Setback Module	6
5	Velocity Change Vs Setback Curve	8
6	Modular Setback Automated Testing Machine	12
7	First Modification	16
8	Second Modification	17
9	Third Modification	19
10	Alternate Lock Down Position	20
11	Fourth Modification	21
12	Lock Down Feature with Die Change	24

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INTRODUCTION

The modular setback pin (MSP) was designed to operate in both the timer and trigger of the M577 mechanical time fuze. The device senses all standard gun firing environments and differentiates between them and handling shocks, including drops of 40 feet and higher. Further, the unit locks in the functioned position (down) after the setback pulse ceases. This prohibits the setback pin from interfering with the operation of the spin detent. Three parts make up the module as shown in fig. 1: two state-of-the-art zinc die castings and a coil spring. Due to the setback module's extremely small size, it may be applied to a variety of other artillery fuzes, taking advantage of its unique operating features. The complete drawings are shown in Appendix A. Also included in Appendix A are additional drawing changes required to accommodate the MSP. Parts required to be modified include the timer spin detent, all timer plates and the trigger spacer.

The primary functional feature of the setback pin is the Z-shaped groove in its side. The groove continues circumferentially around to the far side of the pin, then extends straight down to the bottom of the pin. There is a slot in the top of the pin to allow a screwdriver-type tool to rotate the pin. This pin is hollow to allow the spring to ride inside it.

The main functional feature of the housing is the diamond-shaped cam on its inside surface. Extending upward from the bottom of the housing is a spring support pin which acts as a guide to keep the spring from buckling. There is a section of the mounting flange removed in order to allow the module to be mounted as close to the outside of the M577 timer as possible. There are also two flats on opposite sides of the flange to enable the module to be constrained from rotating by a mating geometry in the plates in which the setback module is mounted. This is shown in figure 2.

The module is assembled by placing one end of the spring in the setback pin and the other end over the spring support pin of the housing. The vertical section of the pin groove is then positioned over the housing cam and the pin is slid into the housing fully. After this operation the pin is rotated one-half turn, this engages the cam with the zig-zag section of the groove as is shown in fig. 3. The pin can then be compressed. As the pin is compressed, it oscillates rotationally.

A computer-aided study to determine the time required for the MSP to function under various setback loads is described in this report. A modified version of this program determined the relative importance of nine design features as they relate to drop height. Limited physical testing was also conducted. A study and conceptual layout of automated assembly and testing equipment for the MSP are also described.

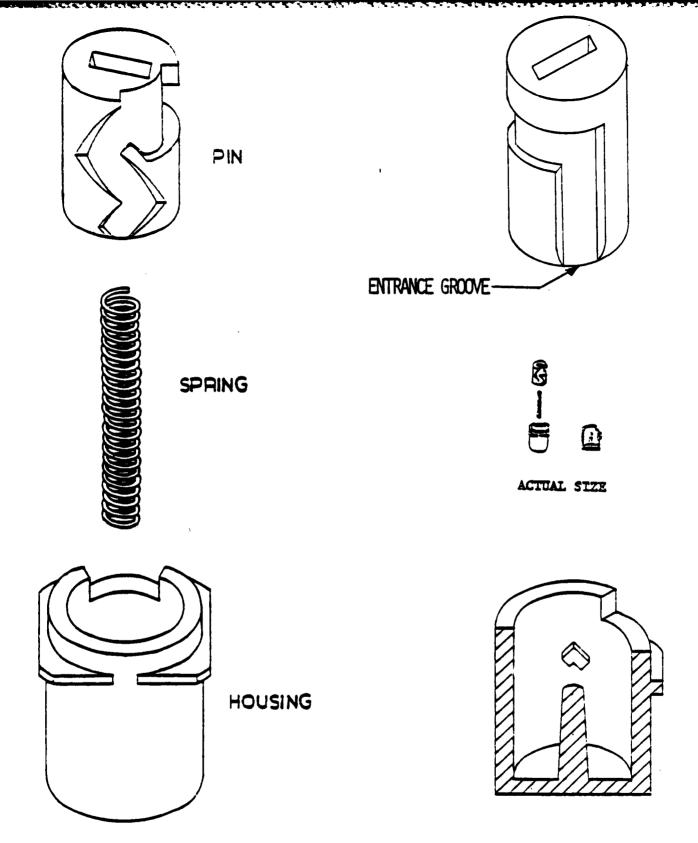


FIGURE | EXPLODED VIEW

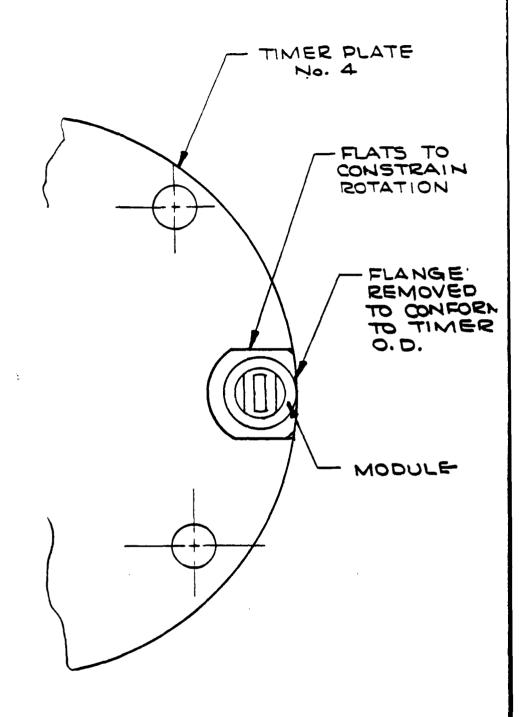
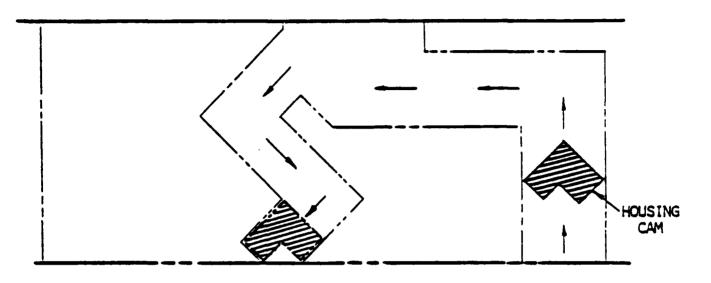


FIGURE 2

NO. 4 PLATE MATING GEOMETRY



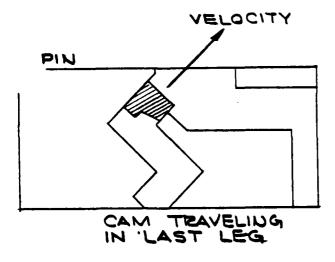
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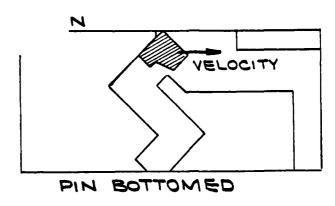
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ASSEMBLY PROCEDURE

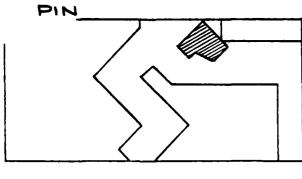
OPERATION

The unique operating feature of the modular setback pin is the rotational oscillation it exhibits while being compressed. Because of this, the MSP functions as a crude runaway escapement, integrating setback and time. As the setback pin is compressed, the housing cam encounters sharp changes in the direction of the pin groove (see fig. 3). Each time the pin groove changes direction, the pin which, of course, is turning in the same direction as the groove, must stop momentarily before turning in the new direction. The pin must then be re-accelerated to continue its travel down the groove. This is analoqous to the motion of the balance (sometimes called the lever) in a runaway escapement. Because the pin does actually stop, the kinetic energy it has when it reaches the end of a leg of the track is expended upon impact of the corner. Therefore, energy must be resupplied to continue motion. This energy is supplied by the setback force. Should the setback force not be present, the pin, instead of continuing to compress, will be returned by its spring to the safe position. Herein lies the key to the pin's ability to differentiate between the setback forces of gun-firing and the forces from handling shocks. Handling shocks have too short a duration to re-accelerate the pin at the beginning of the second and third legs of the Z-groove. A standard setback pin would compress regardless of the duration, as long as the setback force was great enough. Whereas, the MSP must "sense" a combination of setback and duration in order for it to arm.

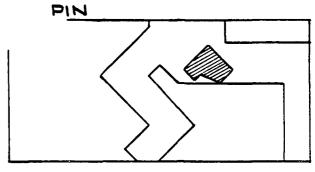
An additional feature of the modular setback pin is its ability to lock down in the functioned position. When the setback pin is completely compressed, it is designed to come to rest directly under a notch in the housing which will catch the pin after cessation of setback (see fig. 4). The MSP will then remain in the down position unless the pin is manually twisted in the counter-clockwise direction. This lockdown feature allows the setback pin to be effectively separated from the spin detent, once it has seen the required setback pulse. Such a feature would also be useful during assembly since many of the testing and assembly operations require that the setback pin be temporarily compressed.











AFTER CESSATION OF SETBACK, PIN TRAPPED

LOCK-DOWN FEATURE OF MODULE

COMPUTER SIMULATION

The equations of motion for the setback pin have been developed by D. L. Overman of Harry Diamond Laboratories and are shown in Appendix B. Due to the complexity of the setback pin motion, certain simplifying assumptions were necessary to reduce the equations of motion to a workable level. These assumptions are as follows:

- o Collisions at the end of each stage are completely inelastic.
- o The spring is linear.

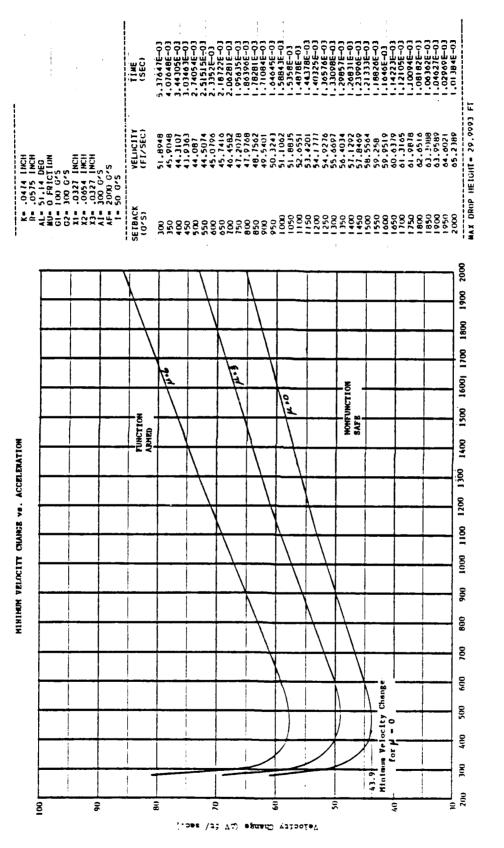
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- o No friction occurs due to side loads.
- The setback pulse is rectangular.

The assumption of a rectangular setback pulse provides a conservative estimate of the maximum safe drop height. Also, the maximum safe drop height is calculated assuming an ideal impact surface. Under all other conditions, the MSP would require a higher drop height in order to arm.

Using these equations of motion, a computer program was written (Appendix C). An evaluation of the parameters used in the program is shown in Appendix D. The quantity computed by the program is the minimum velocity change required to cause the setback pin to function. The computation is iterated for a variety of values of acceleration, generating a curve of minimum velocity change versus acceleration. A typical curve is shown in fig. 5. This process is then further iterated for a variety of friction coefficients, since it is not possible to make an accurate determination for the value of the friction coefficient in this Each curve defines the boundary between a function and a non-function For values of velocity change and acceleration above the curve, the pin will function; and for values below the curve, it will not. Upon examination of the curve shown in fig. 5, it can be seen there is a minimum value of velocity change. For the "no friction" curve in fig. 5, the minimum velocity change is approximately equal to 43.9 ft/sec. An increase in the setback level does not reduce the velocity change required to cause the pin to function. This is due to the fact that the setback duration must also satisfy the MSP before it will function. From this minimum velocity change a drop height can be calculated [Drop Height = $(V_2-V_1)^2/2$]. The drop height for this no friction curve is about 30 ft. This is the minimum height that the MSP can be dropped from in order to cause it to arm. Below this minimum drop height there is no possible impact surface condition that can cause a combination of acceleration and duration great enough to force the unit to function. This is the maximum safe drop height.

This computer program is used to estimate the required time for the MSP to function at various setback loads, thus simulating different weapon systems. Once the design characteristics of the MSP (i.e., spring strength, helix angle, etc.) have been established, the only other factor which affects the time of



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FIGURE 5 CURVES

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setback duration required for the pin to function, is the magnitude of the setback force. Hence, in order to determine the time required for the MSP to function in a particular gun, it is only necessary to know the setback associated with that gun for a particular charge. A list of setback versus duration is shown in Appendix E. The design characteristics used in this program are taken from units used in the final ballistic test conducted in May 1983.

In actual use, the setback duration required to function will vary from 1.3 milliseconds with a 4.2" mortar in Zone 0 to .3 milliseconds in a 105mm Howitzer Zone 7.

A modified version of this program is also used to provide a comparison of the relative importance of nine design features as they relate to the maximum safe drop height (see Appendix F). This information can be very useful in determining more exactly what tolerances are acceptable. This may be important because the current need to chemically etch each part lends itself to large dimensional variations between lots.

AUTOMATED ASSEMBLY

Since the MSP is a self-contained entity capable of being assembled and tested outside the fuze, it is a natural candidate for automation. The following is a conceptual study of automated assembly and testing equipment. The automated assembly portion of this study was conducted in cooperation with Mikron Haesler Limited.

For the purpose of assembling the MSP, Mikron Haesler has proposed using a modified Polyfactor machine, type 90-1-18. The machine is a fully automatic rotary transfer assembly machine comprised of 18 stations, and all feeding, transfer and inspection devices, including laboratory feed bowls and a memory/control system. In addition to normal main machine control, this memory/control system will be used in conjunction with inspection devices located around the machine. The inspection devices are utilized to verify the "condition" of the piece parts as they are introduced to the machine fixtures. Conditions such as presence and position would be "sensed." The information obtained by the individual inspection devices is relayed and accumulated by the memory/control system which, subsequently, instructs the succeeding work, and transfer units, to suspend or continue operations relative to the component piece part condition. The memory section also instructs the ejection stations to separate "good" units from "faulty" or incomplete units. This ability could be utilized to detect broken housing spring supports at Station 12 (see Operation Sequence). This is the biggest part defect we have seen thus far.

All of the component piece parts contained in the assembly, with the exception of the spring, are proposed to be vibratory bowl fed. The spring will be produced by an automatic spring winder at the assembly machine. These springs will be 100% tested immediately after assembly and the reject rate can be recorded by the memory/control. If the reject rate becomes too high, the memory/controller can be programmed to stop operations and wait for adjustments.

The following is the operation sequence for the assembly of the MSP submitted to HTI by Mikron Haesler. The machine is designed to run at a foursecond cycle rate, but this time can be reduced if need be. The price for this Polyfactor assembly machine, including a "Modicon" brand memory/control system and a factory trial run, was quoted at approximately \$247,000.*

Sequence of Operation:

Station	<u>Operation</u>	Equipment
1	Feed and transfer "setback pin" nest (1)	Bowl Feeder
2	Inspect presence and position "setback pin"	
3	Invert "setback pin" if required	
4	Orient "setback pin"	
5	Transfer "setback pin" to nest (2)	

^{* 1983} dollars

Station	<u>Operation</u>	Equipment
6	Inspect presence and position "setback pin"	
7	Vacant	
8	Feed and transfer "spring" nest (2)	Spring Winder
9	Inspect presence and position "spring"	
10	Feed and transfer "housing " nest (3)	Bowl Feeder
11	Orient "housing"	
12	Inspect presence and position "housing"	
13	Transfer "housing" and rotate to lock-in position	
14	Inspect presence, position and movement of "housing"	
15	Vacant	
16	Eject and separate good/bad assemblies	
17	Eject partial assemblies	
18	Inspect empty nests	

After the assembly operations have been completed, the units would automatically be loaded onto a testing machine. HTI has envisioned the testing machine to be compromised of six (6) 10" spin tables all mounted on a 44" diameter rotary table (see fig. 6). The cycle rate for the testing fixture would be geared to the rate of the assembly machine. By doing this, the machine could take advantage of the setback pin positioning already done by the assembly machine. Each of the six spin tables would hold ten setback pins and would be controlled by identical stepper motors mounted underneath the large table. A controller at each of the six divisions of the large index table determines the action of the individual internal spin tables; that is, as the large table indexes, the controller's electrical contacts for each of the six fixed positions is engaged for the individual motor at that position. The motor, which first "homes" into a constant nest position, then makes ten consecutive steps for one revolution, or spins at a preset speed depending on the controller setup.

Setback units are transferred from the assembly machine to the testing machine at station no. 1. The large table then indexes to station no. 2. Setback units that arm at station no. 2 (non-arm spin) are photo-optically detected for lack of pin projection and ejected at station 3. Any units that fail to arm at station 4 are photo-optically detected for pin projection and automatically ejected at station 5. All parts off of station number 6 are good and the spin table is emptied in preparation for reloading at station number 1.

SPECIAL AUTOMATIC TESTING MACHINE MODULAR SETBACK PIN

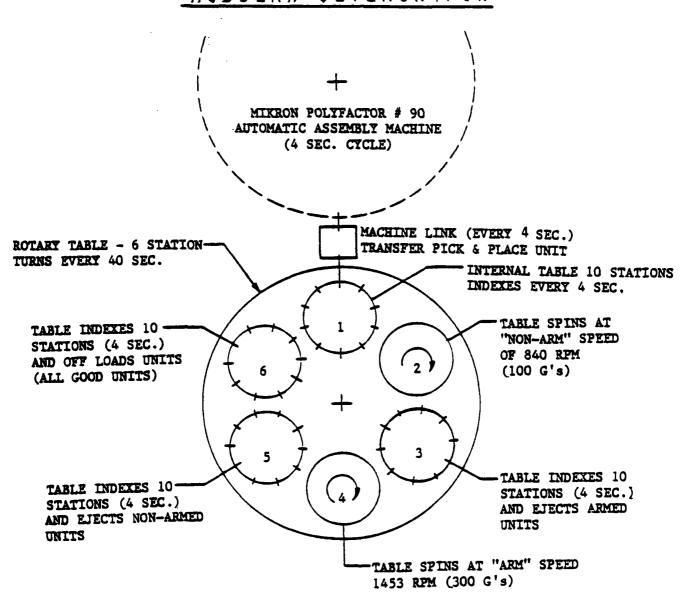


FIGURE 6
AUTOMATIC TESTING MACHINE

Inspections for presence would be performed after each loading and unloading operation. The location of these devices and the transfer units themselves would be on a stationary surface just outside the large index table. It is HTI's estimate that this test machine can be designed, built and installed for approximately \$160,000.*

MANUFACTURABILITY

To manufacture the die castings for the modular setback pin, state-of-theart die-making techniques were involved. A special tool was built to generate the helical-radial sides of the Z-groove and cam. This was used to make a 10:1 scale model of the complex surfaces. From those models, the dies were made by a reducing pantograph.

When the first parts were made from these dies it was found that a serious problem existed in measuring the complex surfaces. A number of methods were considered and all were eliminated as being unfeasible, except for the toolmaker's microscope. Even this proved inadequate, since its repeatability in this application was greater than the tolerance on the dimensions being measured.

Ultimately, the criterion for acceptance of the parts was their ability to interact properly. As cast, the parts would not fit together, the cam being too large for the track - or the track being too small for the cam. The critical dimensions were then adjusted by chemical etching, first by a proprietary process by the manufacturer, then by a final etching of the housing at HTI. This etch schedule is given in Appendix G. Because the etching done by the manufacturer was not tightly controlled by the manufacturer, it was necessary to vary the in-house etch time on a lot-to-lot basis.

The parts used in the final testing had some secondary operations used in their manufacture for reasons of efficiency. Specifically, these were cutting the slot in the housing and cutting the flats on the housing's mounting flange. In production, these features will be cast. It was simply faster to machine them than to change the dies at that point. The same argument applies to the changes to the timer spin detent, which is also a zinc die casting.

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ENGINEERING TEST PROGRAM

A complete modular setback pin ballistic test summary is shown in Appendix H. The first fuzes with modular setback pins were tested in May, 1980, with the results shown in Table 1.

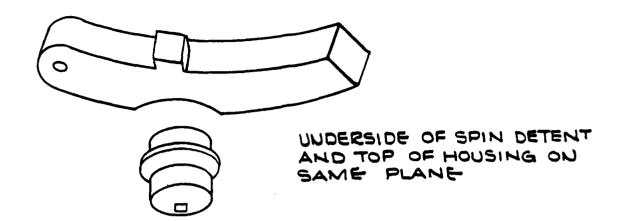
<u>Gun</u>	Zone	Fuze settina	Rounds fired	Rounds functioned
105-mm	7	50 s	5	5
8-in	1	15 s	5	2
155-mm	1	PD	5	5

TABLE 1

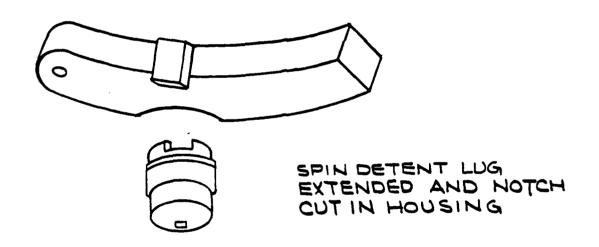
The success in the low zone PD test, compared to the high dud rate in the low zone airburst test, suggested that the problem was in the timer setback pin, and not the setback pin in the trigger. The duds were recovered, and it was found that the timers had either stopped or had never run. The reason for this was not apparent. In the same time period, five fuzes were air gun tested at approximately 30,000 g. Of the ten modules involved, five had the cams sheared off the inside of their housings.

Based on the results, two new tests were planned. Twenty-four fuzes were built for an 8-in., zone 1, vertical recovery test, and twelve fuzes were built for a 105-mm 8, recovery vehicle test. These were fired in August 1980. In all of the high zone tests, the modules functioned properly. The modules were functional afterwards, indicating the cam had sufficient strength for normal gun firing. In the low zone recovery test, four fuzes failed to function. On examination, it was found that the timers had not run. While the reason for this was not apparent, it was found that a modification made to the spin detent, to allow it to pass over the setback pin module, also allowed the balance wheel to be released under some conditions. Modifications to the spin detent and housing, as shown in figure 7, solved this problem. The lug on the spin detent, which engages the blance wheel was extended downward, and the housing was notched to allow the extended lug to pass outward.

Sixteen fuzes of the first modification design were built and fired in recovery vehicles at low zone. On November 1980, twelve fuzes were recovered, and two of these had failed to function. It was found that in the fuzes, which had functioned properly, the pins were retracted and locked down. In the failed units, the pins were fully up. While no reason for this was apparent, it was found that the modules were free to rotate, which could have allowed the housing slot to move out of line with the path of the spin detent lug. To solve this problem, the shape of the housing's mounting flange was altered to have flats, as shown in figure 8. The opening in plate 4 was modified to have a matching shape. It was speculated that another possible cause of the failures was a marginal strength spring, although the springs tested satisfactorily.



ORIGINAL DESIGN FOR MODULAR CONCEPT



FIRST MODIFICATION

FIGURE 7
INTERACTION OF TIMER SPIN
DETENT AND HOUSING

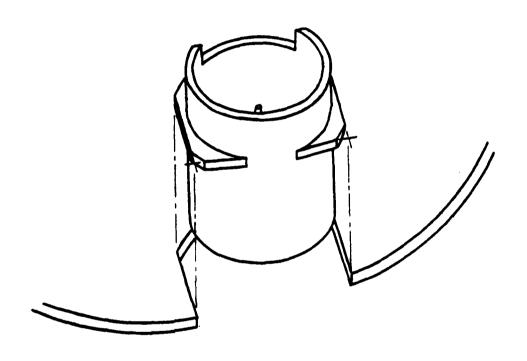


FIGURE 8 SECOND MODIFICATION FLATS ON HOUSING AND PLATE 4 TO PREVENT ROTATION

statically, they could possibly be too strong in a dynamic environment. In further analysis, a comparison was made between the two enclosures in which the modules are placed in the fuze, since all failures had been encountered in timers, and none in triggers.

It was determined, the greatest difference between the two locations was that the timer no. 1 plate forms a wall outward of the pin, as shown in figure 8, and there is no similar feature in the trigger where the pin stands free.

Two groups of twenty test fuzes each were built. Both groups had the flats on the mounting flange. One group had weaker springs, and the other group had material removed from the sidewall of plate no. 1, as shown in figure 9. Eighteen fuzes from each group were fired in recovery vehicles at low zone in March, 1981. The group with the weaker spirngs had a high failure rate. Those with the relief cut in plate no. 1 had two failures of the timer to operate. However, in each of these latter failures, the pins were down and locked, yet the spin detents were still blocked.

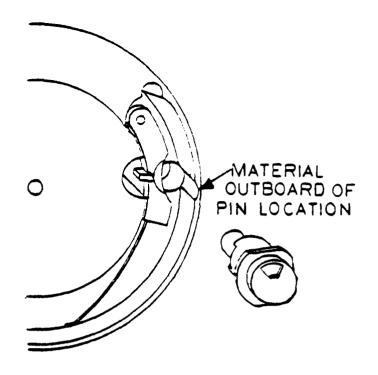
On close examination, it was found that the top of each locked-down pin was slightly above the housing, rather than flush with it. The reason for this is the interaction of the Z-groove and diamond cam, as shown in figure 10. While the locked position was designed to be as shown in the upper view, which puts the top of the pin flush with the top of the housing, the configuration shown in the lower view also provides effective locking, but in this case the pin is approximately .015" above the top of the housing. It was not feasible to lower the module, nor to reduce its size further. Therefore, it was decided to remove some material from the spin detent, allowing it to pass over the pin in either of the locked-down configurations. This is shown in fig. 11.

Thirty fuzes were built using the configuration described above, and including all previous modifications. These were fired in June of 1981, in low zone recovery vehicles. All fuzes were recovered and all were found to have functioned properly.

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Based on these previous results, a larger test program with 200 units was prepared. The test started; however, due to a large number of duds, it had to be stopped prematurely. Thirty recovery rounds were fired to analyze the problem. Fifteen were fired in the 155-mm zone 8, cold, and set for 75 seconds. Four of the fuzes failed to function but, when the setback pins were examined, all were found to have locked down in the timer and all but one were locked down in the trigger. The other fifteen rounds were fired in the 155-mm zone 1, cold, and set for thirty seconds. Only eight of the fifteen rounds functioned; however, there were just two setback pins in the timer that were not locked down, and one in the trigger. So, although the setback pins could not be blamed for the entire problem, there were nevertheless some difficulties with the setback pins. All of the work accomplished up to this point was done on a previous MSP contract, number DAAK10-77-C-0152 (CPFF).

On this latest MSP contract (DAAK10-82-C-0132), a test program involving 121 units was planned. The following changes were performed to these MSP units:



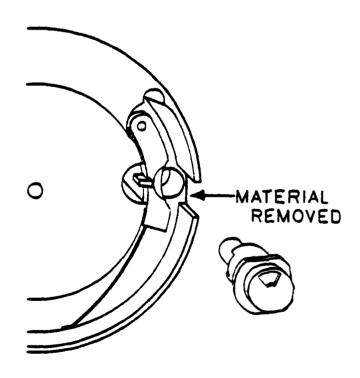
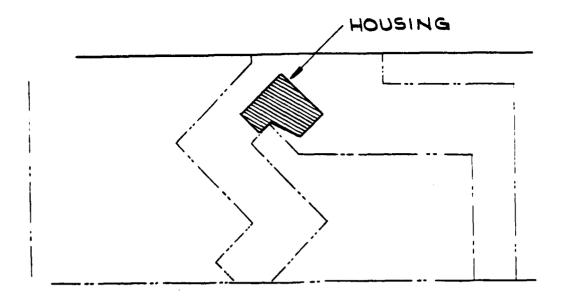
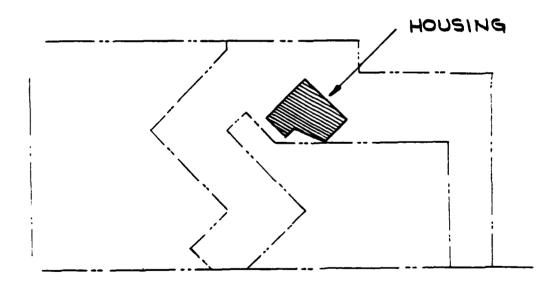


FIGURE &
THIRD MODIFICATION
MATERIAL REMOVED FROM SIDEWALL
OF PLATE NO. 1



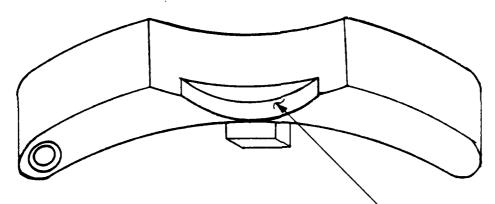
INTENDED LOCK POSITION



ALTERNATE LOCK POSITION

FIGURE 10

RELATIVE LOCATIONS OF Z-TRACK & CAM



MATERIAL REMOVED TO ALLOW CLEARANCE FOR PIN IN ALTERNATE LOCK-DOWN CONFIGURATION

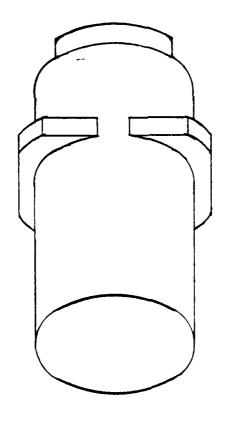


FIGURE 11
FOURTH MODIFICATION

Lighter Springs

All units tested incorporated a lighter spring that required an average of 256g's to compress them. Previously, springs required about 600g's. Although the springs were lighter, these units were still able to pass a 40-foot drop test.

100% Cycling

After a light ϵ tching, all units were cycled at least 70 times. The major problem we have been experiencing with these MSP units is that the cam in the housing is oversized. Cycling reduces this cam size without reducing the size of the rest of the unit as etching does.

100% Centrifuge Testing with a 400g Function Cut-off Point

All units that required more than 400g's to function were rejected. There was no definitive cut-off point in previous tests.

NO Vydax

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The units tested had no lubrication at all. Since Vydax wears away with prolonged friction, it would serve no purpose to have it in these units which were to be cycled 70 times. Also, the Vydax could possibly get gummy in the cold test and do more harm than good.

With the aforementioned changes made, a 121 fuze ballistic test was performed. This latest test was a 100% success. All 242 setback pins in 121 fuzes functioned properly. Even the 8" zone ½, which had two duds in the control group, functioned perfectly. The control group may have experienced a "cross-over" problem. This is a situation in which the setback force subsides before the spin detents have a chance to move out. The setback pin would then come back up locking out the spin detent. This cannot happen with the MSP since it is designed to lock down after setback.

There were three duds in the rough handling group; however, such fuzes are only required to be safe to handle and fire. The X-rays of these rough handling duds did show the modular setback pins in the safe position after rough handling.

The most significant change made to this last test was the 100% cycling. Cycling the units assured that they were capable of moving freely. This would not be necessary if the parts were made to drawing tolerances. The complete success of this latest ballistic test proves the present MSP configuration can be both relaible and safe.

CONCLUSIONS AND RECOMMENDATIONS

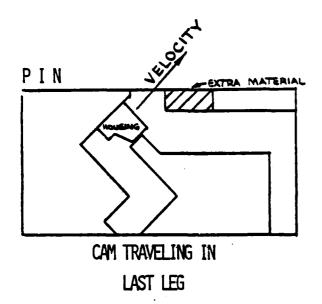
The modular setback pin described in this report has shown the potential to improve the reliabilty of the M577 fuze. The MSP provides excellent performance in all drop test environments and functions reliably in all gun-firing environments. Its ability to lock down after setback can eliminate the "crossover problem." The MSP has the potential to be less expensive to manufacture than the present setback pin systems. The MSP also offers ease of assembly due to its self-contained design; the fact that the same unit can be used in both the trigger and timer and because of its ability to lock down during testing. Also, the MSP can be tested outside the fuze prior to assembly.

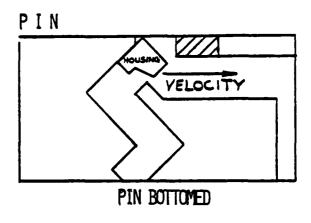
Although the MSP design has proven feasible, an additional engineering effort will be required before it is ready for production. This is due to the extreme complexity of the die-cast housing and pin. The parts presently used do not fit together as cast; they had to be etched in order to reduce the size of the housing cam. Of course, this also reduced the dimensions which were not oversized causing additional problems. Cycling, as was done in the last ballistic test, could eliminate this problem but would be extremely costly for production size quantities. Therefore, to correct this problem, the manufacture of a new Housing die insert will be necessary. However, care must be taken to avoid the problems encountered with the first die.

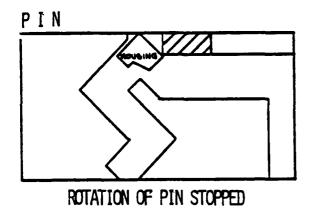
The root of these difficulties is our inability to accurately measure these complex surfaces. We were therefore forced into the undesirable position of accepting parts based on their ability to interact properly. Etching had to be performed on a trial and error basis. This problem of accurate measuring must be addressed before production can be considered practical.

Therefore, it is recommended that additional engineering work be performed to reduce these problems before implementation. Using the computer study supplied in this report, it may be possible to increase some non-critical tolerances and increase the Z-groove width to ease manufacturing and reduce the need for etching. One possible method for the accurate measurement of these die cast parts is to section sample parts in such a manner that critical dimensions could be measured along flat surfaces. This method would not be useful in measuring one particular part, but rather a large group of parts. Die casting generally produces parts with a high degree of consistency so once the dimensions of one part is known it can be assumed that all the parts will be very similar. Reproducibility is exceptionally exact with the state-of-the-art processes used by Gries Dynacast to make the MSP. Other difficulties such as the cam shearing in the 30,000g air gun test and the breaking of the housing spring supports could also be avoided. This may be accomplished by going to a stronger material such as Zamak No. 5. The problems caused by allowing two lock down positions could be eliminated with a minor die change that would force the cam into its original lock down configuration (see figure 12). Also, the need for a cutout on the no. 1 plate should be further investigated.

The MSP design has proven to be beneficial in terms of reliability; but, due to the manufacturing difficulties, it is not being recommended for implementation at this time.







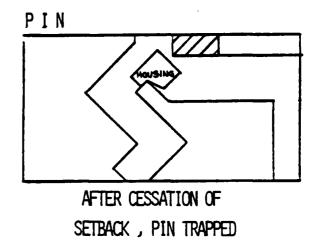
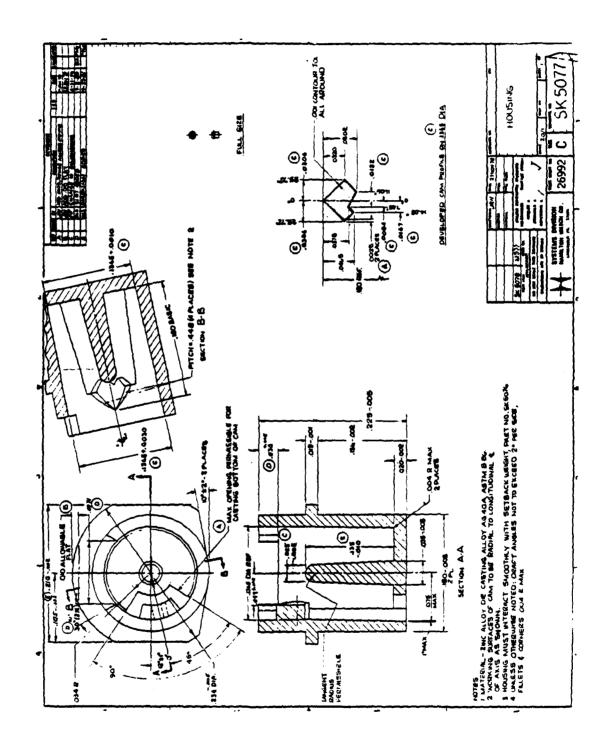
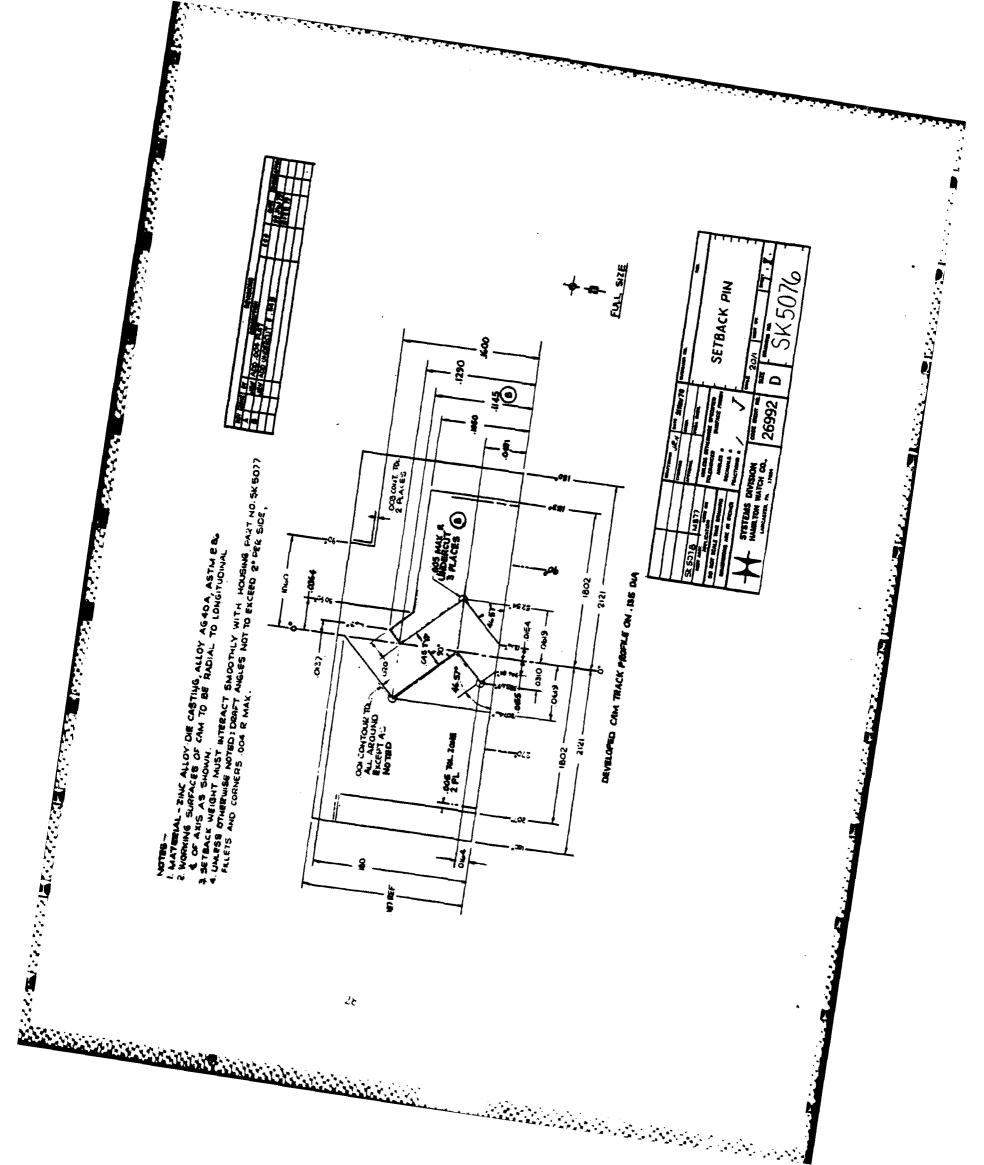
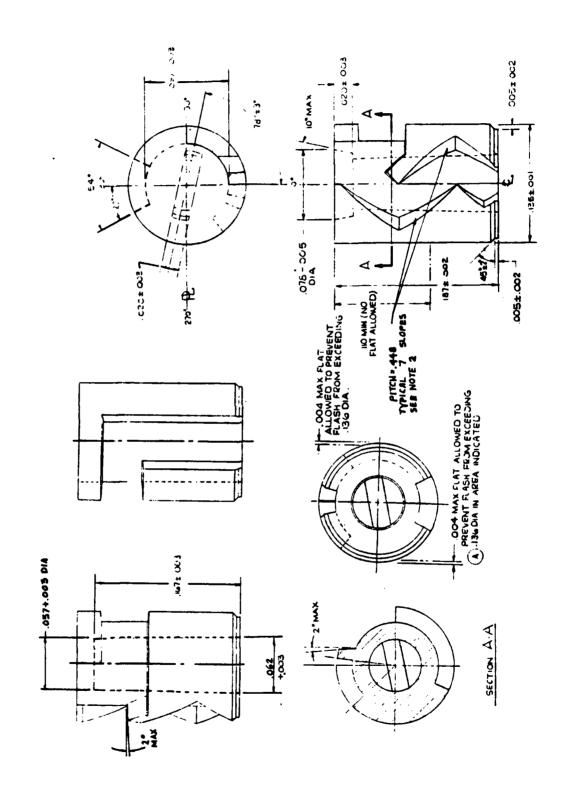


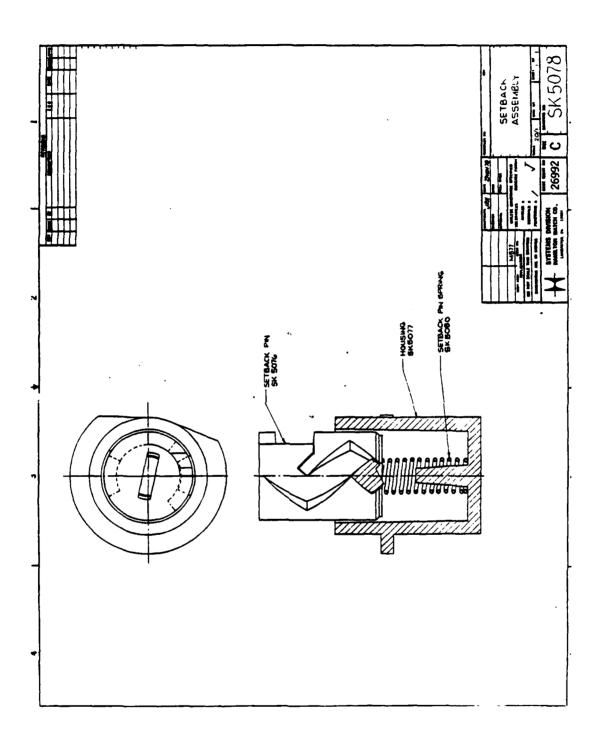
FIGURE #12
LOCK-DOWN FEATURE WITH DIE CHANGE

APPENDIX A REQUIRED DRAWING CHANGES









8 SPRING PATE 1-3-79 UMLESS OTHER TOLERANCES ANGLES & 2 MEXT ABEV. UNED ON SPRING MUST OPERATE FREELY IN . 057 DIA BORE 164 LENGTH .296 LENGTH ٠. WIRE SIZE - OOG DIA TOTAL NO. OF TURNS -25 STYLE OF ENDS - I CLOSED ON EACH END LOADS- .119 LB. MIN. (54 GM.) @ .228 LB. MAX (103 GM.) @ MATERIAL - CRES 302 DIRECTION OF WIND - LEFT HAND ADVISORY INFORMATION AND OVER . 035 DIA PIN OUTSIDE DIA - .050 FREE LENGTH-.420 SPRING DATA 94 4 m - ai m 8

31

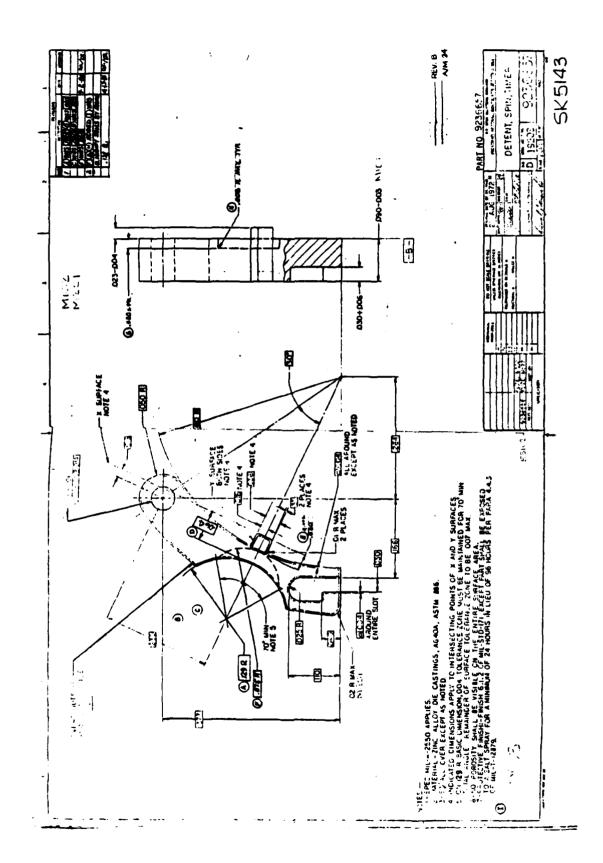
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DECIMALS &

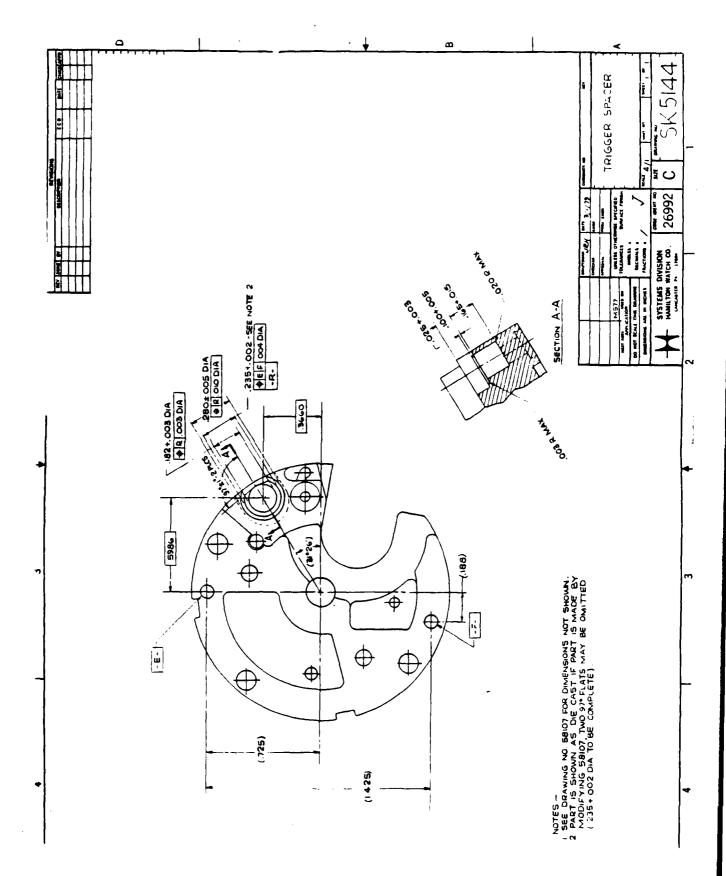
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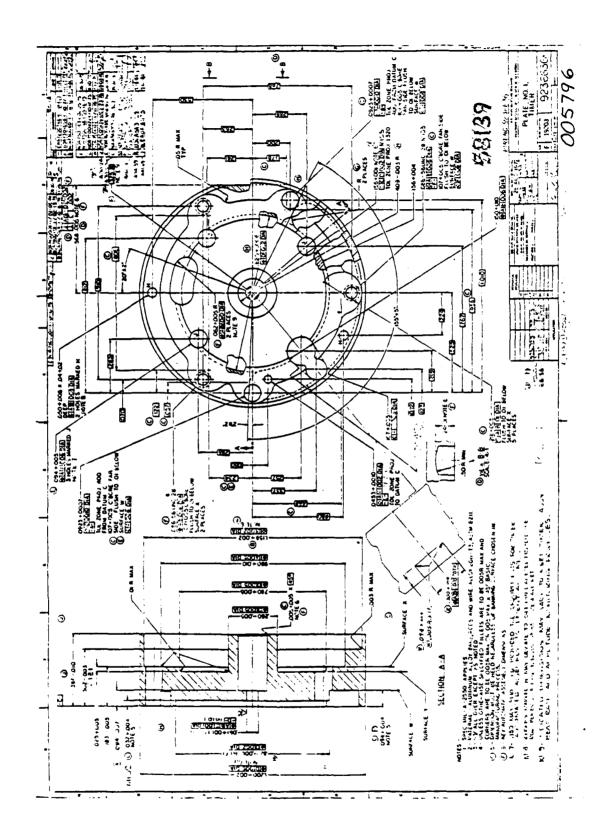
26992

SYSTEMS DIVISION HAMILTON WATCH CO., LANCASTER, PA. 17604

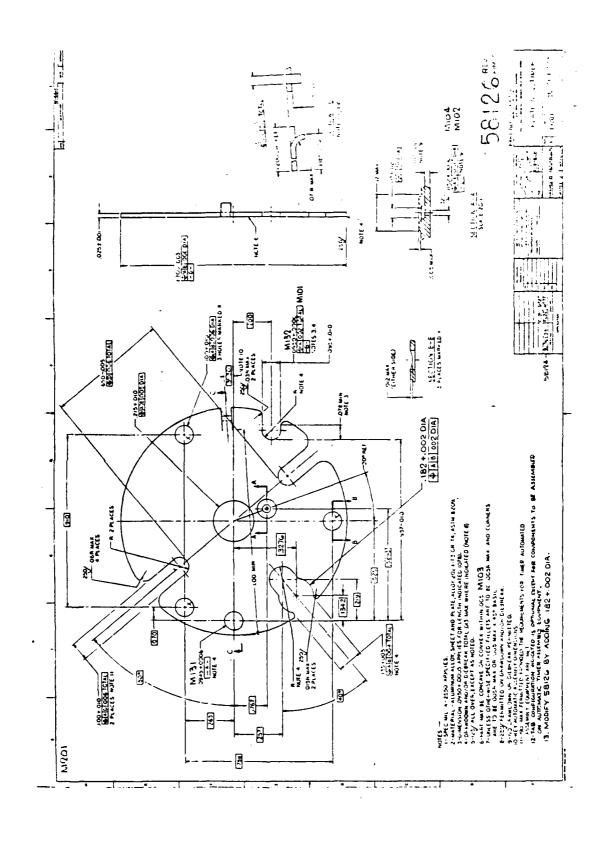


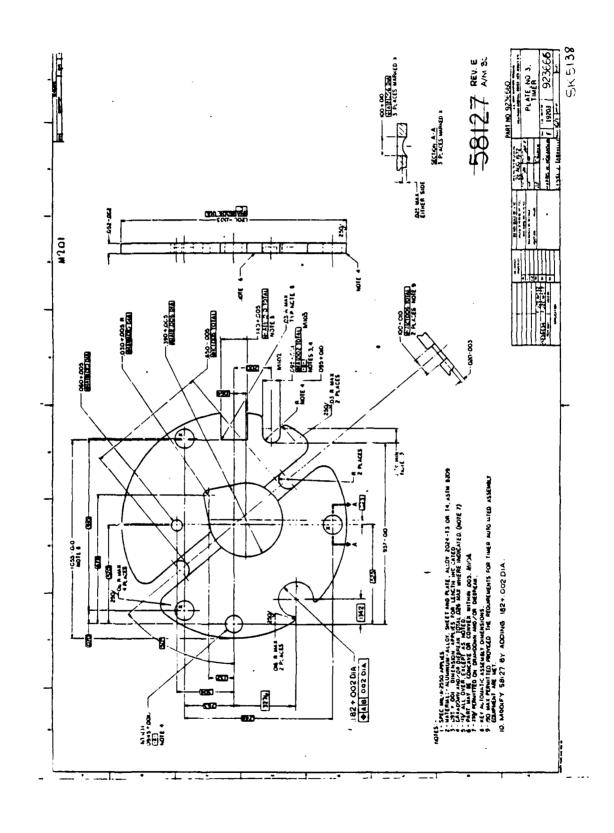
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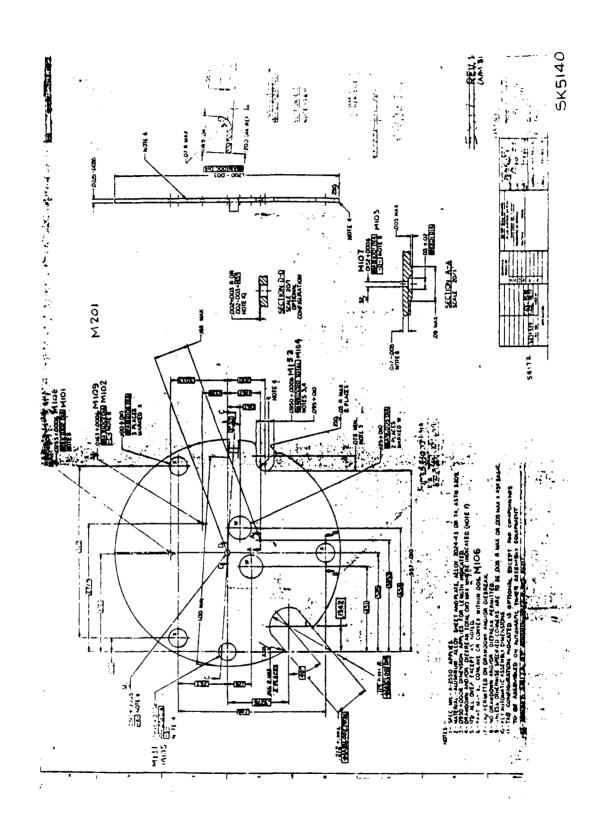


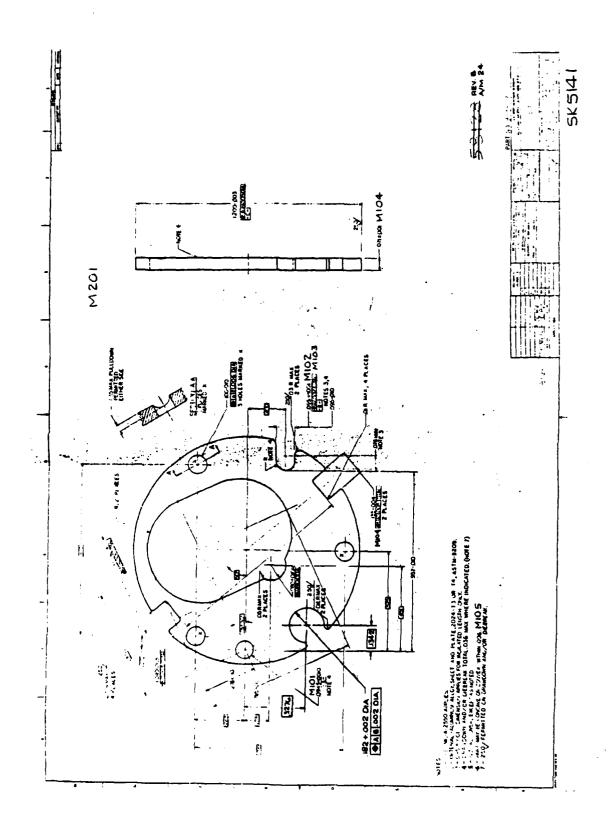


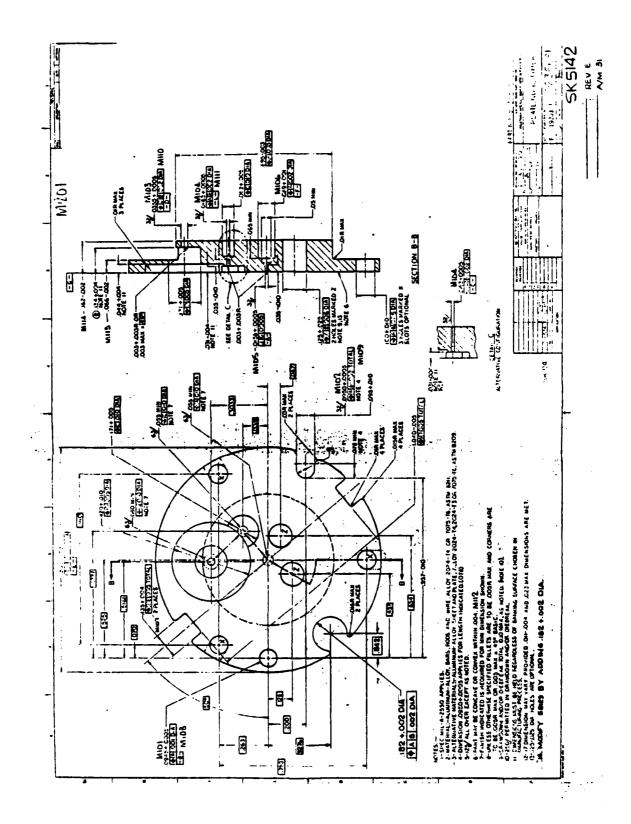
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APPENDIX B

MODULAR SETBACK PIN

EQUATIONS OF MOTION

M577 Modular Setback Pin

Minimum Velocity Change for Function - Computer Program Equations from D. L. Overman (Oct. 73)

$$V_{\min} = \sum_{i=1}^{f} V_{i}$$

$$V_{i} = A \sqrt{\frac{g K_{i}}{B}} \cos^{-1} \left[1 - \frac{F \Delta^{X_{i}}}{A - G_{1}} - \sum_{i=1}^{i-1} \Delta^{X_{i}} \right]$$

$$K_{i} = 1 + \left(\frac{k}{r}\right)^{2} \qquad \boxed{\frac{1 + \mu \tan \alpha_{i}}{\tan \alpha_{i} (\tan \alpha_{i} - \mu)}}$$

$$\tan \alpha_i = \frac{L_i}{2\pi r}$$

$$B = \frac{G_2 - G_1}{\sum_{i=1}^{f} A_{x_i}}$$

Variables DLO Comp. Prog

ULU	comp. Frog.		
k	K	radius of gyration	inches
r	R	radius of interaction	inches
L		helix lead	in/turn
	AL	helix lead angle	degrees
μ	MU	coefficient of friction	_
G	G1	spring bias, safe position	g¹s
G2	G2	spring bias, armed position	g's
Δx_i		length of i th leg	inches
•	X1	length of first leg	inches
	X2	length of second leg	inches
	Х3	length of third leg	inches
Α		driving acceleration	g's
	ΑI	driving acceleration, first iteration	gʻs
	AF	driving acceleration, final iteration	gʻs
	I	increment of iteration	ğ's

F = 1 for all legs except final

$$F = \frac{G_2 - .5 B\Delta \chi_F}{A}$$

APPENDIX C MODULAR SETBACK PIN COMPUTER SIMULATION PROGRAM

APPENDIX C MODULAR SETBACK PIN COMPUTER SIMULATION

```
10 DEF FNACS(X)=ATN(SQR((!-X^2)/X^2))
20 DEF FNNCS(X)=(P1/2-FNACS(X))+P1/2
30 PI=3.141592654#
40 READ K,R,A4,G1,G2,X1,X2,X3,M1,A5,A6,I
50 DATA .0474..0575.51.14.138.247..0327..0654..0327..3.300.2000.25
70 INPUT #AI=#:45
80 INPUT #AF=#:46
90 INPUT # I=#:I
100 PRINT
                K=";K;"INCH"
R=";R;"INCH"
110 PRINT®
120 PRINT"
               AL=" : A4 : " DEG"
130 PRINT#
               MU=" IMI I "FRICTION"
140 PRINT
               G1=#1G1 +#G/S#
150 PRINT"
160 PRINT"
               G2=#1G21#G/S#
170 PRINT#
               X1="1X1;" INCH"
               X2=" | X2 | " INCH"
180 PRINT
               X3=" + X3 + " INCH"
190 PRINT#
200 PRINT
               AI="; A5; "G/S"
210 PRINT
               AF=" 1 A61 "G S"
                I=# : [ : #G / S#
220 PRINT
230 PRINT*-
240 PRINT"SETBACK
                              TIME
250 PRINT* (G*S)
                              (SEC)#
200 A4=A4*2*P1/360
270 X=TAN(A4)
280 IF MI>X THEN PRINT "TOO MUCH FRICTION"
290 K1=1+(K/R)^2+((1+M1+X)/(X+(X-M1)))
300 S9=X1+X2+X3
310 B=(G2-G1)/S9
320 S8=SQR(386.088*K1/B)
330 F1=1
340 SI=0
350 S2=X1
360 S3=X1+X2
370 FOR A=A5 TO A6 STEP I
380 F3=(G2-_5*B*X3)/A
390 A1=1-F1+X1/((A-G1)/B-S1)
400 A2=1-F1+X2/((A-G1)/B-S2)
410 A3=1-F3=X3/((A-G1)/B-S3)
420 Z1=ABS(A1)
430 Z2=ABS(A2)
440 Z3=ABS(A3)
450 IF Z1>1 OR Z2>1 OR Z3>1 THEN P3=1*GOTO 630
460 IF AI=0 THEN BI=PI/2
470 IF A2=0 THEN B2=PI/2
480 IF A3=0 THEN B3=PI/2
490 IF AI<O THEN BI=FNNCS(AI)
500 IF A2<0 THEN B2=FNNCS(A2) ...
510 IF A3<0 THEN B3=FNNCS(A3)
520 IF AI>O THEN BI=FNACS(AI)
530 IF A2>0 THEN B2=FNACS(A2)...
540 IF A3>0 THEN B3=FNACS(A3)
550 VI =A+S8+BI
560 V2=A+58+82
570 V3=A+S8+B3
580 V=(V1+V2+V3)/12
590 IF A=A5 THEN V9=V
600 IF V<V9 THEN V9=V*H=I
610 IF V>V9 AND H=1 THEN H=2
620 T=V/(A+32.174)
630 IF P3=1 THEN PRINT At#
                                     PIN WILL NOT ARM WITH GIVEN DATA"
640 IF P3=0 THEN PRINT A,V,T
650 P3=0
660 NEXT A
670 PRINT"-
680 D9=.5*V9^2/32.174
690 IF H=2 THEN PRINT MAX SAFE DROP HEIGHT = 109 + "FT"
700 IF H<>2 THEN PRINTMINSUFFICIENT RANGE TO DETERMINE MAX SAFE DROP HEIGHT"
710 END
```

APPENDIX D
EVALUATION OF COMPUTER
SIMULATION PARAMETERS

Appendix D Evaluation of Parameters used in Computer Program

Radius of gyration is defined by:

$$I = \kappa^2 M$$

and

 $I = \int r^2 dm$

where

I = moment of inertia

K = radius of gyration

M = mass of pin

dm = element of mass

r = distance from axis to element of mass

The moment of inertia of a complex body can be computed by breaking it into smaller bodies and adding the moments of inertia of its components, thus:

$$I = \Sigma I_{i}$$

The pin is divided into the following component bodies:

- 1. Main cylinder
- 2. Internal cylinder
- 3. Screwdriver slot
- 4. Entry groove
- 5. Circumferential groove
- 6. Zig-zag groove
- 7. Bottom groove

The moment of inertia of each is computed by:

1. Main cylinder

$$I_1 = 1/2 \text{ M } r^2$$

$$M_1 = \rho \pi r^2 \ell$$

where:

r = radius of cylinder .0675 in. ℓ = length of cylinder .187 in. ρ = density of zinc (experimentally determined) .22 lb./in.³ I_1 = 1.341 x 10⁻⁶ lb. in.² M_1 = 5.889 x 10⁻⁴ lb.

2. Internal cylinder

r = .032 in. $\ell = .167$ in. $M_2 = 1.182 \times 10^{-4}$ lb. $I_2 = 6.052 \times 10^{-8}$ lb. in.²

3. Screwdriver slot

 $M_3 = \rho 2 l h b$ $I_3 = M(\frac{\ell^2}{3} \frac{h^2}{12})$ $\ell = \text{half of slot length .0365}$ h = width of slot .020 b = height of slot .020 $M_3 = 6.424 \times 10^{-6} 1 b$ $M_3 = 3.067 \times 10^{-9} 1 b$ in.²

4. Entry groove

 $I_4 = f r^2 dm$

 $M_4 = f dm$

dm = proh dr

M₄ = ρθh/r dr

 $I_4 = pensr^3 dr$

 $M_4 = 1/2 \rho \theta h r_2^2 - r_1^2$

 $I_4 = 1/4 \rho\theta h r_2^4 - r_1^4$

dm = element of mass

r = radius to dm

dr = thickness of dm

h = height of groove .160 in.

0 = angle subtended by groove .945 radians

 r_1 = inside radius of groove .048 in.

 r_2 = outside radius of groove .068 in.

 $M_4 = 3.858 \times 10^{-5} \text{ lb.}$

 $I_4 = 1.337 \times 10^{-7}$ lb. in.²

5. Circumferential groove

$$M_5 = 1/2 \rho \theta h r_2^4 - r_1^2$$

$$I_5 = 1/4 \rho\theta h r_2^4 - r_1^4$$

 θ = 2.466 radians

h = .055 in.

 $M_5 = 3.461 \times 10^{-5} \text{ lb.}$

 $I_5 = 1.199 \times 10^{-7}$ lb. in.²

6. Zig-zag groove

$$M_6 = 1/2 \rho \theta h r_2^2 - r_1^2$$

$$I_6 = 1/4 \rho\theta h r_2^4 - r_1^4$$

$$\theta = .917 \text{ radians}$$

$$M_6 = 4.376 \times 10^{-5}$$
 lb.

$$I_6 = 1.516 \times 10^{-7}$$
 lb. in.²

7. Bottom groove

$$M_7 = \rho \pi h (r_2^2 - r_1^2)$$

$$I_7 = 1/2 M_7 (r_2^2 + r_1^2)$$

$$h = .0075 in.$$

$$r_1 = .063 in.$$

$$r_2 = .068 in.$$

$$M_7 = 3.395 \times 10^{-6} \text{ lb.}$$

$$I_7 = 1.458 \times 10^{-8}$$
 lb. in.²

The total mass is:

$$M = M_1 - M_2 - M_3 - M_4 - M_5 - M_6 - M_7$$

$$M = 3.439 \times 10^{-4} \text{ lb.}$$

The total moment of inertia is:

$$I = I_1 - I_2 - I_3 - I_4 - I_5 - I_6 - I_7$$

$$I = 8.576 \times 10^{-7}$$
 lb. in. ²

The radius of gyration, k, is determined by:

$$\kappa = (\frac{I}{M})^{1/2}$$

$$K = .0499 in.$$

Radius of interaction

The radius of interaction is approximated as occurring at half of the depth of the groove, or .058 inches.

Helix Lead Angle

The helix lead angle is taken at the radius of interaction.

$$AL = tan^{-1} (a/c)$$

where:

AL = helix lead angle

a = axial displacement = .0654 in. for second leg

c = circumferential displacement = .0532 in. at .058 in. radius

 $AL = 50.87^{\circ}$, or .888 radians

Spring forces

The spring is specified as:

.119 lb. at .296 in. .228 lb. at .164 in.

The spring rate is therefore:

R = -.8258 lb./in.

and the spring equation is:

$$F = -.8258 L + .3634$$

Starting and finishing spring lengths are:

 $L_1 = .2986 in.$

 $L_2 = .1715 in.$

So starting and finishing spring forces are:

 $F_1 = .1168 \text{ lb.}$

 $F_2 = .2218 \text{ lb.}$

Since the pin weight was determined to be 3.341 x 10^{-4} lb, the starting and finishing loads in g's are:

$$G_1 = 350 g$$

$$G_2 = 664 g$$

APPENDIX E

SETBACK MAGNITUDE

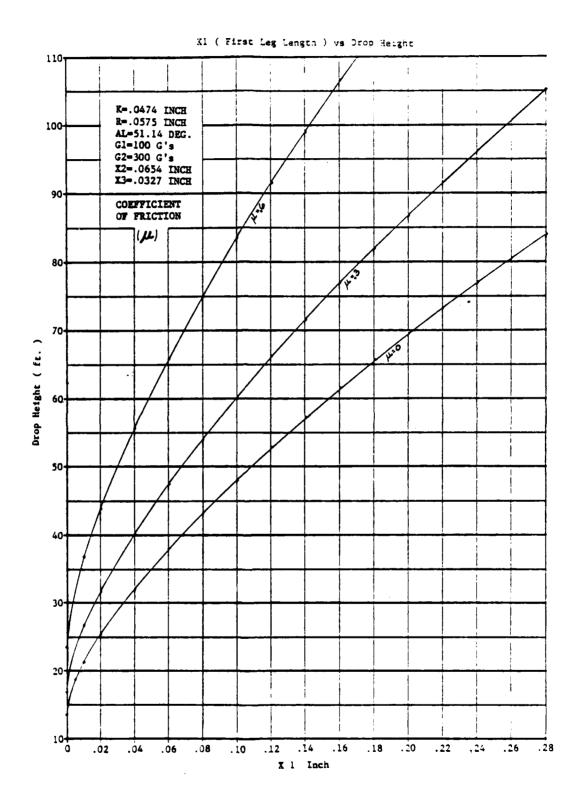
VERSUS

TIME REQUIRED FOR

MSP TO FUNCTION

2.81525E-04 2.80811E-04	2.8U267E-04	2.79705E-04	2.79044E-04	2 720625-04	2.77328E-04	2,7679E-04	2.76257E-04	2.7561E-04	2. /5087E-04	2.74351E+U4	2 73364E-04	2,72860E-04	2.72216E-04	2.71707E-04	2.71.21.2E-04	2,705/35-04	2 605055-04	2. 60087F-04	2.08458E-04	2.67979E-04	2.67492E-04	2.0/0!4E-04	2.0055E-04	2.05921E-04	2.640H5F-04	2.64514E-04	.63907E	2.63456E-04	2.0/983E=04 2.6/983E=04	2 010415-04	51 496E	2.61039E-04	2.00584E-04	2.599/E=04 2.595/E=04	2.500HQF-04	2.58639E-04	58055E	576291	2.5/208E -04	56 339E	.55771E	2.55347E-04	2,54925E-04	54503E	2.5 59.39E-04	2.53538E-04	7000							
24400	24600	24700	24800	25,000	25100	25200	₹300	25400	25500	25,000	25800	25900	2,6000	26100	26200	20700	26500	2000	26700	26800	26900	27000	27100	27200	22400	27500	27600	2700	2,600	28000	28100	28200	28300	28500	28600	28700	26800	28900	20100	20200	29300	29400	27500	246(0	25.700	00000	3000							
3,3128E-04 3,30326E-04	3.29327E-04	3.28365E-04	3.2/4/25-04	3.255255-04	3.24669E-04	3.23742E-04	3,22787E-04	3,21,8805-04	3 200535-04	3.102436-04	3.183546-04	3.17461E-04	3.16679E-04	3.157896-04	3.150246-04	3,132676-04	3,124036-04	3.116555-04	3.108146-04	3.09994E-04	3,09257E-04	3.084025-04	3.0404.6.04	3.06018F-04	3.053076-04	3.04525E-04	3,03831E-04	3,03029E-04	3.01547E-04	3.00728E-04	3.000655-04	2.99278E-04	2,98632E-04	2.97204E-04	2.90441E-04	2.95796E-04	2.950446-04	2.944046-04	2.930265-04	2,924E-04	2.91 064E-04	2.910496-04	2.90294E-04	2.89682E-04	2 863481-04	2.877685-04	2.870535-04	2.86465E-04	2.85754E-04	2.65183E-04	2 630136 04		2.82647E-04	2.8208E-04
17900	18100	18200	8300	18500	18600	00ZR1	1 88 00	0068	200	0000	000	9400	19500	00961	888	200	5000	20100	20200	20300	20400	888	0000	20800	20002	21000	21100	21200	21400	21500	21600	21700	21800	2200	22100	22200	22300	22400	22600	22700	22800	22900	23000	23300	00116	23400	23500	23600	23700	23800	2000	24100	24200	24300
4.21024E-04	4.17118E-04	4.15225E-04	4.132936-04	4.11.454E-104	4.07808E-04	4.06026E-04	4.04254E-04	4.02488E-04	4 00108E-04	3. 97379E-04	3.95776E-04	3.94104E-04	3.92503E-04	3,90876E-04	3 973636 04	3.86240F-04	3.84659E-04	3.83166E-04	3.8167E-04	3.80201E-04	3.7875E-04	3 758625-04	3.150025-04	3.730526-04	3.717246-04	3,70351E-04	3.68969E-04	3.6/6!5E-04	3. A5009F-04	3.6375E-04	3.6244E-04	3.61219E-04	3.59926E-04	3.5741E-04	3.56269E-04	3.55E-04	3.53843E-04	3,52010E-U4	3.50335E-04	3.49198E-04	3.48084E-04	3.47002E-04	3.45801E-UA	3.44716104	3.43634E-04	3.41506E-04	3.40428E-04	3.39296E-04	3.38263E-04	3.37237E-04	3.35.05.04	3.3427E-04	3.33274E-04	3,32258E-04
11400	00911	11700	0081	200	12100	12200	.12300	12400	2000	12200	12800	12900	13000	13100	3200	35	3500	3600	13700	13800	3800	38	38	38	004	14500	14600	88	200	2000	15100	15200	15300	25.0	15600	15700	15800	0005	00191	16200	16300	16400	00501	000	38	0000	7000	00171	17200	1300	200	2009	0027	1 7800
6.60269E-04 6.53001E-04	6.45962E-04	6.391 77E-04	6.32566E-04	0.20131E-04	6.13874E-04	6.07982E-04	6.02244E-04	5.96703E-04	5.91275E-04	5.808565-04	5.75818F-04	5.70943E-04	5.66148E-04	5,615236-04	5.50992E-04	5, 48217F-04	5.43967E-04	5,39813E-04	5,35765E-04	5.31 79E-04	5.27912E-04	#01941446 #02586	5 14753G-04	5,131958-04	5.09683E-04	5.0623E-04	5.02911E-04	4.99587E-04			ш	뿞		4.75201E-04		.!.	뜅	4.64398E-04		56678	삤	4	200	40912E	4226AF	40036E	*	35581E	蝉	4.31264E-04	4 2706 3E-04	щ	23023E	
5000	5200	5300	5400	2000 0000 0000	5700	5800	2800	0009	38	9300	0400	6500	0099	007.0	0800	2000	80.2	7200	7300	7400	200	822	3 5	2002	8000	8100	8200	9700	8500	8600	8700	9800	0000	80	9500	9300	9400	6,500	200	0086	0066	0000	38	0000	200	10500	00901	00701	00801	885	3	1500	11300	
*************************	0474		2.1	MONE OF PRICE LINE	62= 243 6'S	XI = .0327 INCH	X2= .0654 INCH	X3= .0327 INCH	AI = 300 G/S	8	****			w.	-ī r	2,57217E	2,27655E	i	-	-	1.000	1.47383F	1.40476F	1,34413E	1.29039E	1.24233E	1800	1.1245-0	•		1,0330SE	_ ,	2500 9.82989E-04	, .	9.1948IE	2900 9,00805E-04	3000 8.831 7/E-04	33.00 8.003076=04	3 33	.21492E	60 f	3500 7.94983E-04	-,-	3900 7 59487E-04	4000 7.48659E-04	,-		4300 7,18626E-04		4500 4 01838-04 1	4700 6.83546E-04	6.7552BE	6.61769E	

APPENDIX F A RELATIVE COMPARISON OF NINE DESIGN FEATURES AS THEY RELATE TO DROP HEIGHT



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ABBERRATERER REPORTER KAN	N= .0575 INCH	AL= 51.14 DEG	MU- O FRICTION	CI* 100 G'S	C2= 300 G'S	XI = .0327 INCH	X2= .0654 INCH	X3= .0327 INCH	A1= 301 G/S	AF= 2000 G'S	5.9 1 =1	******

		091.0				0.180		061.0	0.195	0.200	0.205	0.210	0.215	0.220	0.225	0.230	0.235	0.240	0.245	0.250	0.255	0.260	0.265	0.270	0.275	0.280	0.285	0.290	0.295	0,300	_
DROP HEIGHT (FT)	13,683	18.829	21.416	23.592	25.555	27.378	29.100	30.743	32,322	33.846	35,324	36.761	38.162	39.529	40.867	42.177	43,463	44.725	45.965	47.185	48.386	49.570	50. 736.	51.88/	53.023	54.144	55.252	56.348	57.431	58.502	59.562
(XI) (IN)	0.00	0.005	0.010	0.015	0.020	0.025	0.030	0.035	0.040	0.045	0.050	0.055	0.000	0.065	0.00	0.075	0.080	0.085	0%0	0.095	0.1.0	0.105	0.110	0.115	0.120	0.125		0.135	0.140	_	0.150

60.611 64.080 64.080 64.100 65.712 66.705 66.038 70.601 71.550 71.550 71.550 71.550 71.550 71.550 71.650 71

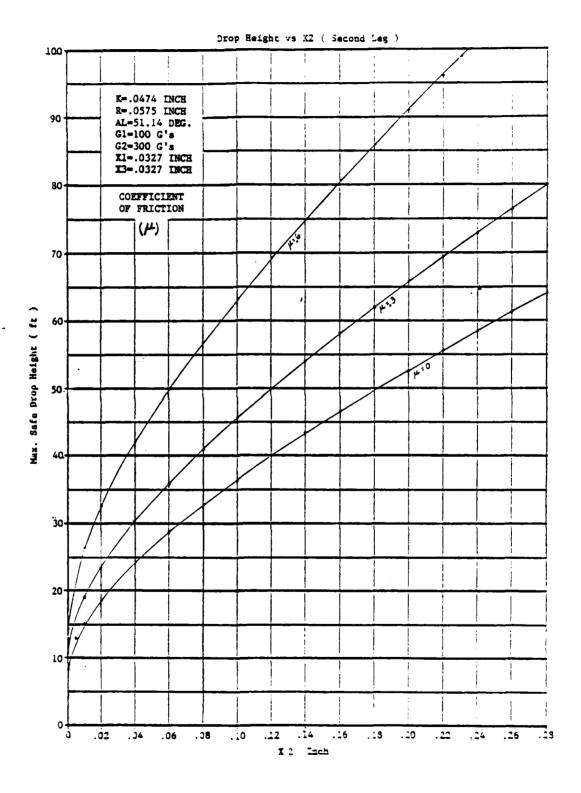
***********	0474 INCH	.0575 INCH	51.14 DEG	.3 FRICTION	100 0/5	300 G/S	.0327 INCH	.0654 INCH	.0327 INCH	301 645	2000 G/S	1 G/S
****	*	.	AL.	# O#	5	* 70	<u>-</u>	X2=	×3=	-	AF.	

	7	0.165	~	-	٦.	_	•				•			0.225											•	•	•	•	•			
DROP HEIGHT (FT)	17.075	23.496	26.725	۲.	31.890		•	38,364	۳.		•	•	•	49.329	•	•			•	•		•	•	۲.	791.99	67.567	68.949	70.316	71.668		.32	¢
CXI) CINI	0.000	0.005	0.010	0.015	0.020	0.025	0.030	0.035	0.040	0.045	0.050	0.055	0.000	0.065	0.070	6.0.0	0.080	0.085	0.000	0	0.100	_	0.110	0.115	0.120	0.125	0.130	0.135	-	: -	Ξ,	0.155

76.933 78.218 79.490 80.752 82.005 84.471 86.905 86.907 86.207 92.825 92.825 92.825 92.825 93.134 96.207 100.790 101.902 105.200 105.200 105.200 106.287

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	AL= 51.14 DEG	NO6 FRICTION OI = 100 0/S	02= 300 6/5	XI 0327 INCII	X20654 INCH	X3* .0327 INCH	AI= 301 G/S	AF= 2000 G/S	I= 1 G'S

	104,725	106.520	108.298	110.060	111.806	113,538	115.254	116.957	118.646	120,322	121.985	123,636	125,275	126,903	128.519	130,125	131,721	133,306	134,882	136.448	138,004	139,552	141.091	142,621	144.143	145.057	147,163	148.661	150,152	151.635	1
	0.155	0,160	0.165	0.170	0.175	0,180	0.185	0.190	0.195	0.200	0.205	0.210	0.215	0.220	0.225	0.230	0.235	0.240	0.245	0.250	0.255	0.260	0.265	0.270	0.275	0.280	0.285	0,290	0.205	0,300	
DROP HETONIT (FT)	23-641	32,533	37.003	40.763	44.154	47,304	50,279	53, 118	55.846	58,480	61,034	63.516	65.036	68,299	70.610	72.875	75,005	17.276	79.419	81.527	83.602	٠	87.663	89.65	619.19	93.551	95.466	97.358	99.229		102.912
(XI) (IN)	0.00	0.005	0.010	0.015	0.020	0.025	0.030	0.035	0.040	0.045	0.050	0.055	0.00	0.065	0.070	0.075	0.080	0.085	0.000	0.095	0.100	0.105	0.110	0.115			0.130	0.135	0.140	0.145	0.150



	DROP HELGIT (FT)	8,401	ó	٦.	17.213	~ ب	12	۰,	24,314	•	•	27.780		31.940					•	38.446	٦.	40.180	A 03	٠.	43.525	44, 339	45.144
K= .0474 INCH R= .0575 INCH AL= 51.14 DEG MU** O FRICTION GI = (10 O'S G2** 300 G'S XI= .0327 INCH X3** .0327 INCH X3** .0327 INCH X4** 30327 INCH X5** .0064 INCH X5** .0067 INCH X1** .007 G'S AF** 2000 G'S I= 1 G'S	(X2) (IN)	00.00	8	<u> </u>	0.00 0.00	9	8	g	Š	•	•	0.00 6.000	 •	0.075	0.000	0.00	0.095	•	•	•			27.0			0.145	. 15

45.942 47.231 47.291 49.065 49.065 49.065 51.589 51.589 52.031 52.031 55.739 55.738 55.738 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018 60.018

0.155 0.160 0.165 0.175 0.175 0.180 0.200 0.200 0.225 0.240 0.250 0.250 0.270 0.070

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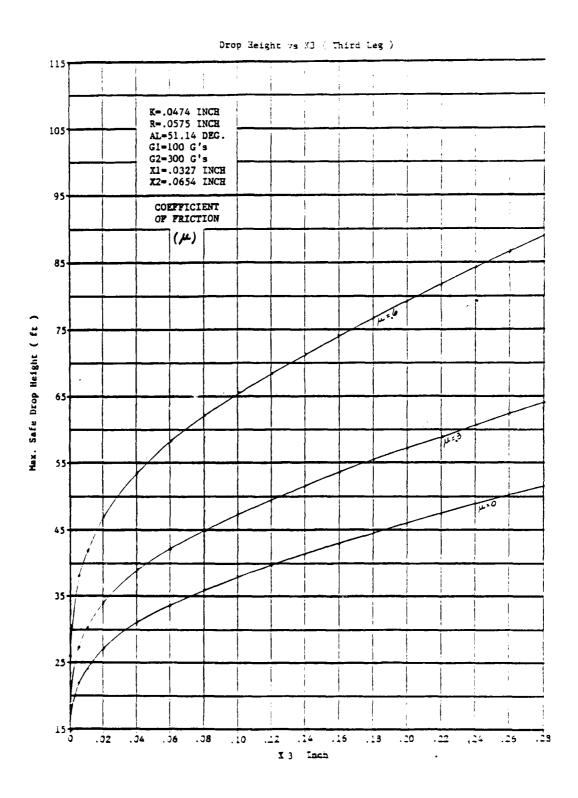
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| K = .0474 | INCH |
| R = .0575 | INCH |
| AL = 51.14 | DEC |
| Al = .0675 | INCH |
| X2 = .0654 | INCH |
| X3 = .0654 | INCH |
| X4 = .0054 | INCH |
| X5 = .0654 | INCH |
| X5 = .0654 | INCH |
| X6 = .0000 | O+5 |
| A = .000
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55.331 59.260 61.260 61.270 63.183 62.183 64.173 65.031 65.031 65.031 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.478 70.200 71.403 71

		0.155	091.0	0.165	971.0	5.00		061.0	0,195	0.200	0.205	0.210	0.215	0.720	0.230	0,235	0.240	0.245	0.250	0.255	0.260	وهر ه	0/7:0	0.275	087.0	0.603	0.205	0.300	3	
NCH NCH NCH NCH NCH SCH SCH SCH SCH SCH SCH SCH SCH SCH S	DROP HEIGHT (FT)	14,616	22.596	26.524	.29-741.	32.578	101.00	39.840	42.010	44.080	46.072	47.098	49.866	51.684	55.103	56.881	58.541	60.171	61.773	63.348	64,899	61 034	61.936	100 OF	70.594	201.51	75, 203	16.600		
K = .0474 INCH H = .0575 INCH AL = 51.14 DEG AUN = .6 FRICTION C2 = 300 G*S XI = .0327 INCH X2 = .0654 INCH X3 = .0327 INCH AL = 301 G*S AF = 2000 G*S AF = 2000 G*S	_	00.00		0.010	0.015	0.020	0.025	0.030	600	0.045	0.050	0.055	0,060	0.065	0,00	0.080	0,085	060.0	0.095	∹	0.105	Ξ,	٠.	٦.	∹ -	7	0.135	: -	0.150	

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	.0575	∿	Ξ	300 6.5	.0327		.0327		2000 G'S	l G/S	*****
*	<u>.</u>	AL-	<u>.</u>	02	÷	X2=	* C ×	A !-	AF.	<u>.</u>	****

	42.602	42.991	43.375	43.756	44.134	44.509	44.880	45.249	45.615	45.979	46.340	46.699	47.056	47.410	47.763	48.114	48.462	48.009	49.154	49.498	40.840	50.160	50.519	50.856	51.192	51.526	51.859	52,191	52.521	52.851	
	0.155	0.160	0.165	0.1.0	0.175	0.180	0.185	0.190	0.195	0.500	0.205	0.210	0.215	0.220	0.225	.0.230	0.235	0.240	0.245	0.250	0.255	0.260	0.265	0.270	0.275	0.280	0.285	0.290	0.295	0.300	
DROP HETGHT (FT)	15.046	21.078	- 24.358	26.048	27.393	28.524	29.510	30, 389	31.188	31.924	32.610	33,254	33.865	34.446	35.003	35.540	36.058	36.560	37.048	37.523	37.088	38.442	38.888	39,325	39.755	40.178	40.595	41.006	41.412	41.814	42.210
(X3) (IN)	000.0	0.005	0.010	0.015	0.020	0.025	0.030	0.035	0.040	0.045	0.050	0.055	0.060	0.065	0.00	0.075	0.000	0.085	0.000	0.095	0.1.0	0.105	0.1.0	0.115	0.120	0.125	0.130	0.135	0.140		0.150

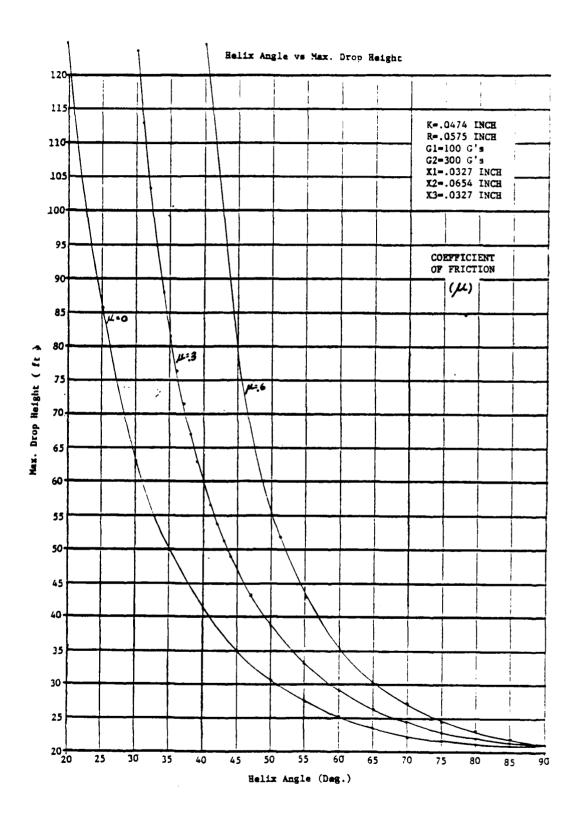
		0.155	-	0.170	Ξ.	=:	0.195	0.200	0.205	0.210	0.250	0,225	0.230	0.235	٠,	~	0.255	'n٠	0.230	~	?	۲.	٠į٠	0.300	•
	DROP HEIGHT (FT)	18.776		- 4	ı.	Φ.	37,923		۰.	4,	42.280	. 0	Τ,	٥, ٧	46.232	₽.	47.405	۱ خ	עֿ כ	9	-	9.0	51.172	- · ·	67
K= .0474 INCH R= .0575 INCH AL= 51.14 DEG AU= .3 FRICTION G1= 100 G'S G2= 300 G'S X1= .0327 INCH X3= .0327 INCH X3= .0327 INCH X1= 301 G'S AF= 2000 G'S I= 1 G'S	(NI) (EX)	0,000	0.010	0.015	0.025	0.030	0.035	0,045	0.050	0.055	0.00	0.000	0.075	0.080	060.0	0.095	=:	=:	0.1.0		-	-:	=:	0.140	0.150

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4 4 4	DROP HEIGHT (FT)	25.007 42.085 45.006 45.006 45.006 45.006 52.506 53.887 56.343 56.343 56.343 56.343 66.635 66.835 66.835 66.835 66.835 67.946 67.946 68.689 68.689 69.420 67.946 69.420 67.946 69.420 67.946 69.420 67.946 69.420 67.946
K** .0474 INCH A!* 50.14 DEG A!* 51.14 DEG AU** 6 FRICTION GI** 100 G'S X!** .0327 INCH X2** .0054 INCH X2** .0054 INCH X3** .301 G'S AF** 2000 G'S I** 1 G'S	Panagananananananananananananananananana	0.000 0.015

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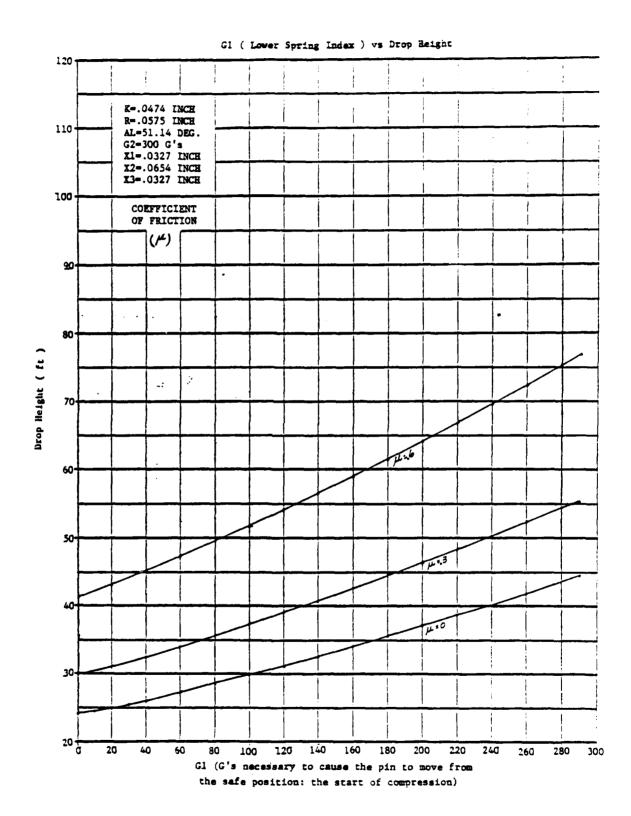
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***************	K= .0474 INCH	R- , 0575 SWCH	S	WU O FRICTION	CI- 100 G'S	02= 300 0'S	XI= .0327 INCII		AF= 2000 G'S	l= 1 G/S	化化物物 化二甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基甲基	

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DROP HEIGHT (FT)	29,006	29.996	29.996	20.996	29.996	29.996	20.006	20.996	29.996	20.096	20,006	20.00	20.006	20.000	20.006	29.006	29.996	20.996	20.00	29.996	29.990	29.996	29.00	20.00	29.00	50.00	30 004	30 004	30 004	200.00	30 004	900 00	29.996	29,996	20.996	29.996	29.996	29.996	29.996	20.996	29.996	29,996	29.996	20.000
(AL) IDEGI		2.000	000	4.000	000	900.9	7.000	000.8	00.6	000.01	000.11	12.000	13.000	14.000	15,000	16,000	17.000	18.000	19,000	20.000	21.000	22.000	23.000	24.000	25,000	26.000	27.000	20.000	29.000	30.00	31.00	32.00	33.00	25.00	36.00	37.000	38.000	30,000	40,000	700.00	42,000	43,000	44.000	45, ((3))

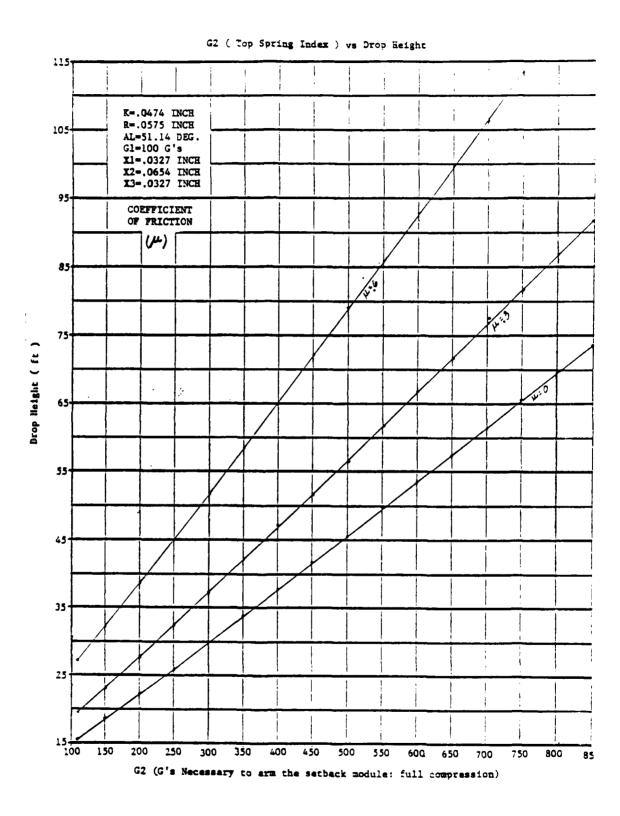
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	UROP HEIGHT LFT		• •		₹,	37.432	Τ.	٦.	•		•	7,7		₹.	•	•	₹.	•	•	•		٠,	•	•	٠,	• •	•	٠,٠		•	٦.	•	•		37.432
.0174 .0174 .0174 .0174 .0174 .0032 .0054 .0054 .0054 .0054 .0054	(AL) 1DE01	000.1	2.08	00.4	5,000	6.000) 7.000	8,000	000.4	200-	200.21	13.000	24.000	000-91	17.000		20,000	21.000	23.000	24.000	%. %.	27.000	28.000	30.05	31.000	32.000	34,000	35.000	36.000	37. W.	39.000	40,000	1,000		44.CX)	45,000

		46.000	47.000	40.000	50.000	51.000	53.000	54.000	56.000	57.000	58.000	00.00	000.19	62.00	64.000	00.59		68.000	69.000	000.17	72.000	73.00		•	78.000	79.000	90.00		83,000	84.000		67.000		90.000
	DROP HETOMIT (FT)	51.4823	51.827	51.827	51.827		: _:	:	-:-	: -:	<u>:</u> .	51.827	_:			_	_	_	51.827		_	-	_	_	51.827	-		: _:		-	51.827	_	51.827	51.827
K= .0474 INCH AL= 51.14 INCH AL= 51.14 INCH INU- 6 FRICTION GI= 100 G-S C2= 300 G-S X1= .0327 INCH X2= .0327 INCH X3= .0327 INCH X1= 301 G-S AF= 2000 G-S I= 1 G-S	(AL) LDEG1	1.000	2.000	900.	2*000	6,000 7,000	9,00			12.000	13.000	14.000	16.000	•	19.000	20.000	21.000	23.000	24.000	26.000	27.000	28.000	30.000	31,000	33.000		•	37.000	•	•	000.	•	43.000	44,000 45,000



The control of the	R= .0474 INCH R= .0474 INCH R= .0474 INCH R= .0475 INCH GI= 1070 G/S (127- 300 G/S (127- 300 G/S A1= 301 G/S A1= 3027 INCH X2= .0654 INCH X2= .0664 INCH X2= .0666 INC
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K = .0474 INCH R = .0575 INCH F = 4 PEC					= G/S
K= .0474 INCH	AU= 0 FRICTION	G2= 300 G'S	X2= .0654 INCH	A[= 301 G/S	la G'S
F= .0575 INCH	GI= 100 G/S	XI= .0327 INCH	X3= .0327 INCH	AF= 2000 G/S	

	000.014	520.000	530.000	540,000	550,000	560.000	570,000	580,000	590.000	000.009	610.000	620,000	630,000	640.000	650,000	000.009	670.000	690.000	690,000	700,000	710.000	720.000	730.000	740.000	750.000	760,000	00.077	780.000	290° (MO	800.000	810.000	820.000	830.000	840.000	850.000	860.000	870.000	880.000	890.000	000*006
DROP HEIGHT LFT.1	911.91	•	17.180	17.894	18.616	19.346	20.082	20.823	21.570	22,321	23.076	23.835	24.597	25.362	26.129	26.899	27.670	28,444	29.219	29.996	30.774	31.553	32,334.	33.116	33.898	34.682	35.466	36.251	37.037	37.823	38.41	39,398	40.186	40.075	41.764	42.554	43.344	44.134	44.925	45.716
(62) (6/5)	110.000	120.000	1 30.000	140.000	150.000	160.000	170.000	180.000	190.000	200.000	210.000	220,000	230.000	240.000	250.000	260. OXD	270,000	280,000	290,000	300.000	310.000	320.000	330.000	340,000	350,000	360.000	370,000	380,000	390.000	400.00	410.000	420.000	430.00t	440.000	450.0XX	460.000	470,000	480,000	490.000	500.000

46.508 477.299 48.609 49.616 50.469 50.469 53.642 53.642 55.236 56.024 56.024 66.336 66.336 66.336 66.336 66.336 66.336 66.336 67.71 77.72

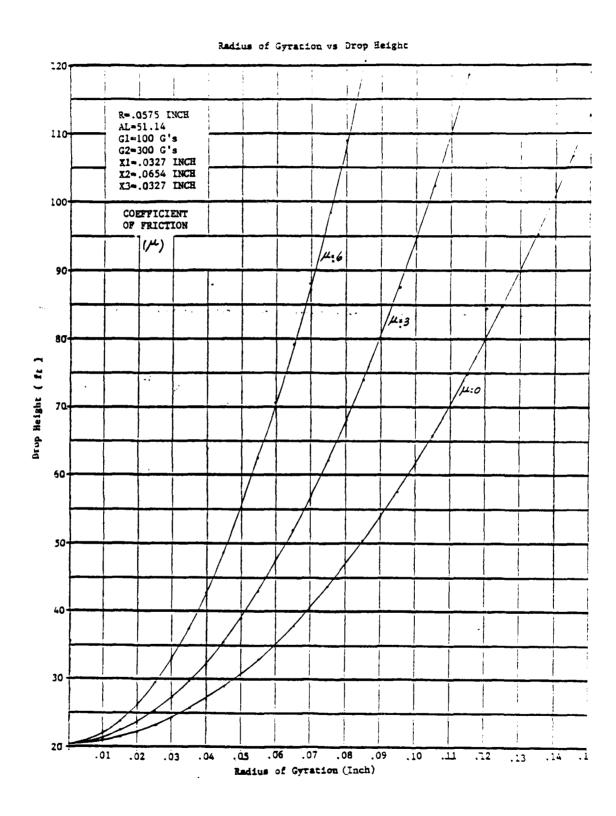
.0474 INCH	.0575 INCH	51.14 DEG	. J FRICTION	100 6/5	300 6/5	.0327 INCH		.0327 INCH	301 6.5	2000 675	1 6/5
<u>"</u>	=	¥Ľ.	3	"	62	= ×	<u>.</u> .	X3=	V	AF=	_

	58.037	50.05	60.013	61.002	166.19	62.980	63.970	64.050	65.949	00.040	066.79	68.921	69.912	70.903	71.895	72.886	13.878	74.870	75.802	76.854	77.846	18.839	19.831	80.824	118.18	82.810	83.803	84.196	85.790	86.783	87, 176	88.770	89.764	90.158	91.752		93, 139		95.158	96.122
	510,000	520,000	530,000	540,000	550,000	560.000	570,000	\$80,000	590,000	000.009	000.019	620.000	630.000	640,000	020.000	000.099	670.000	680.000	690.000	700,000	710.000	720.000	730,000	740.000	750.000	760,000	770.000	780.000	790.000	800.000	810.000	820.000	830.000	840.000	850.000	860.000	870.00	880.000	890,000	900,000
DROP HELGIG (FT)	19.691	20,559	21.439	22,330	23.231	24.142	25.060	25.085	26.917	27.855	28.797	29.744	30.695	31.649	32.607	33.567	34.530	35.495	36.462	37.432	38.403	39.375	40.349	41.325	42,301	43.279	44.258	45.238	40.218	47.200	40.102	27.07		51.12	52.118	53.103	54.089	55.075	56.062	51.049
(62) (6/5)	110,000	120.000	130,000	140.000	150.000	000.001	170.000	180.000	000.061	200,000	210.000	220.0no	230,000	240,000	250.000	260.000	270,000	280,000	290,000	300.000	310,000	320,000	330,000	340,000	350,000	360,000	370.000	380.000	390,000	400.000	410.000	420.000	430.00	440.000	450.(XX)	460. (KK)	470.000	480.000	490.000	500,000

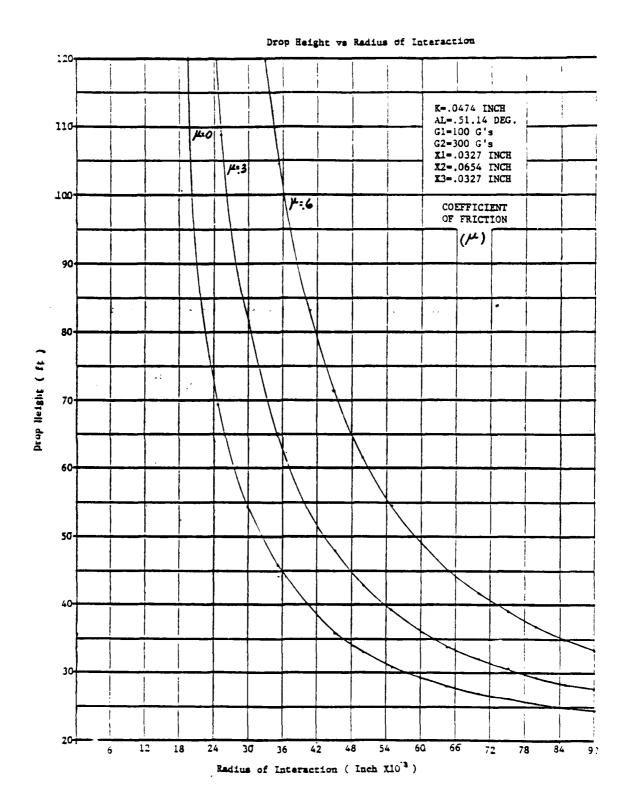
	DROP HEIGHT (FI)	27,264	ĠΟ			38.567														٠,	16.256	ç	78.789 80.156
K* .0024 INCH R* .0034 INCH R* .0034 INCH AL* .0 FRICTION GI* ION G'S G2= 10N G'S K1= .0032 INCH X2= .0054 INCH X3= .0027 INCH X1= 301 G'S AF= 2000 G'S I* I G'S	(62) (6/5)	110,000	140.000	150.000	 000.000		-	240.000		270.000	 	310,000	 	350,000		390.930 900.930	 	430,000	44(), (XX)		 ⁻•	490,000	000.003 000.014

81.724 83.093 84.462 85.831 87.201 89.941 91.202 92.683 95.426 95.426 96.171 99.544 98.171 99.544 98.171 99.544 100.216 100.216 100.216 111.907 111.907 111.907 111.907 111.907 111.907 111.907 112.909 121.533 122.909 121.533 122.909 121.533 122.909 123.909

520,000 540,000 550,000 560,000 600,000 600,000 610,000 640,000 640,000 640,000 640,000 640,000 770,000 770,000 770,000 770,000 800,000 810,000 860,000 860,000 860,000 860,000



MU= U FRICTION G1= 100 G75 G2= 300 G75 X1= .0327 INCH X2= .0327 INCH A1= 301 G75 AF= 2000 G75 I= I G75	110 G/S 110 G/S	AL= 51.14 DEC MU= .3 FRICTION 01= 100 G/S G2= 300 G/S X1= .0327 INCH X2= .0654 INCH X3= .0327 INCH X1= 301 G/S AF= 2000 G/S 1= 1 G/S		K	
(N) (B)	DROP HEIGHT LFT1	(R) (IN)	DROP HELGHT [F1]	CR) CINI	DROP NEIGHT (FT)
0,00	1235.200	0.005	2318,600	0.005	4122,380
0.010	324,409	0.010	570.259	0.010	1046.210
0.015	155.744	0.015	265.012	0.015	476.543
0.020	96.712	0.020	158.175	0.020	277.161
0.025	69.389	0.025	108.725	0.025	184,876
0.030	54.546	0.030	81.863	0.030	134,746
0.035	45.591	0.035	65.666	0.035	104,519
0.040	39.788	0.040	55.154	0.040	84.900
0.045	35.806	0.045	47.946	0.045	71.450
0.050	32,957	0.050	. 42.791	0.050	61.829
0.055	30.850	0.055	38.977	0.055	54.711
0.000	29.247	090.0	36.076	0.000	49.296
0.065	27.999	0.065	33.818	0.065	45.083
0.070	27.009	0.000	32.027	0.070	41.740
0.072	70.411	0.0/2	30.581	5/0.0	19.043
0.030	25,557	0.080	29.308	0.080	36.835
0.085	25.015	0.085	28.418	0.085	35.006
0.000	24.561	0.000	27.597	060.0	. 33.472
0.095	24.177	0.095	26.901	0.095	32,175



AL* 51,14 DEG MU* 0 FRICTION GI* 100 GS G2* 100 GS X1* .0327 INCH X3* .0327 INCH X1* 301 GS AF* 2000 GS I* 1 G*S	.0575 INCH 51.14 DEG 0 0 FRLCTION 100 G/S .0327 INCH .0327 INCH .0327 INCH .0327 INCH .0327 INCH .032 INCH .032 INCH .032 INCH .032 INCH .032 INCH .032 INCH	AL= 51.14 DE0 AU= 3 FRICTION G1= 100 G'S G2= 300 G'S X1= .0327 INCH X2= .0654 INCH X3= .0327 INCH X1= 301 G'S AF= 2000 G'S I= 1 G'S		K= .0474 INCH R= .05/5 INCH AL= 51.14 UEG AU= .0 FRICTION GI= 100 G/S G2= 300 G/S X1= .0327 INCH X2= .0654 INCH X3= .0327 INCH AT= 301 G/S AF= 2000 G/S I= 1 G/S	· 经股份股份股份股份股份股份股份股份股份股份股份股份股份股份股份股份股份股份股份
(K) (IN)	DROP HELGHIT (FT.)	CK) CINI	DROP HEIGHT (FT)	(K) (IN)	DROP HETGHT LFT1
0.005	20,916	9000	20, 098	500.0	21.158
0.010	21.222	0.010	. 21,553	0.0.0	22.194
0.015	21,733	0.015	22.478	0.015	23.919
0.020	22.448	0.020	23.772	0.020	26.335
0.00	23, 368	0.000	23.430	050.0	11.512
0.030	24.492	0.035	20.B74	0.035	31,723
0.010	27,353	0.040	32.648	0.040	42.899
0.045	29,089	0.045	35.791	0.045	48.766
0.050	31,031	0.050	39,305	0.050	55,323
0.055	33.176	0.055	43.188	0.055	62.570
0.0%	15.526	28.5	47.441	0.000	10.50
0.005	30.081	0.065	52.064	0.065	79.134
0.070	40.840	0.070	57.057	0,0,0	88.452
0.075	43.703	0.075	62.419	0.075	98.459
0.080	46.970	090.0	68.152	090.0	109.157
0.085	50,342	0.085	74.254	0.085	120,545
0.090	53,918	0.00	80.726	060.0	132.624
0.095	57,698	0.095	87.568	0.095	145,392

APPENDIX G
ETCHING PROCEDURE
FOR HOUSING

M577 Modular Setback Pin

Etching procedure for housings - to reduce cam size

Process reduces housing OD from .1775" to .1764"

Bright Dip Solution ref: Metal Finishing Handbook 1981, Page 182 263g. Chromic Acid 23g. Sodium Sulphate 1000 ml. D.I. Water room temperature

Zn. Cleaner
Patclin #357 Zinc Cleaner
45 g/1
150°F

Procedure:

- 1. Degrease in chlorothene; dry.
- 2. Agitate in bright dip for 60 seconds; rinse.
- 3. Remove smut in Zinc cleaner in ultrasonic cleaner, approx. 3 mins.
- 4. Rinse and dry.

APPENDIX H
M577 MODULAR SETBACK PIN
BALLISTIC TESTING SUMMARY

M577 MODULAR SETBACK PIN BALLISTIC TESTING SUMMARY

UNITS THAT COULD BE EVALUATED TIMERS TRIGGERS FUNCTIONED FUNCTIONED	5/5 5/5 5/5	24/24 8/8	ı	1	1 1	ı	15/15 7/7 10/10	24/24 14/15 14/15	20/20 20/20 15/15	16/16	15/15
UNITS THAT CO TIMERS FUNCTIONED	5/5 - 2/5	20/24 8/8	10/12	20/20	16/18 11/18	30/30	15/15 7/7 8/8	24/24 15/15 13/15	20/20 20/20 15/15	13/13	15/15
FUZES FUNCTIONED	5/5 5/5 2/5	20/2 4 8/8	10/12	20/20	16/18 11/18	30/30	15/20 7/10 6/10	24/32 11/15 8/15	20/20 20/20 15/15	13/16	15/15
GUN & ZONE	105mm 27 155mm 21 PD 8-inch 21	8" 21 105mm 28	155mm Z1	8-inch Z1	155mm Z1 155mm Z1	155mm Z1	155mm Z8 105mm Z7 8-inch Z1	105mm 27 155mm 28 155mm 21	155mm Z8 8-inch Z1 8-inch Z½	105mm Z7	155mm Z1 (Recovery)
DATE SHIPPED	21 May 80	15 July 80	6 Oct. 80	7 Oct. 80	10 Dec. 80	14 May 81	Sept. 81	(Rough Handling) (Recovery) (Recovery)	May 83	(Rough Handling)	May 83
TOTAL FUZES TESTED	15	32	12	20	36	30	102		106	<u> </u>	15
HAT LOT NUMBER	E022	E028	E031	E030	E044 E045	E051	E056		E107 *		E113 *

^{*} For more complete information on this test see Page

ENGINEERING TEST

SUPP #4-TPR-2594 (YPG)*

WEAPON	ZONE	TEMP	SET TIME	REL	×	4	LPD
(2) 155MM, M198	8(M203)	+1450F	105.0	20/20	105,312	,335	1
(1) 155MM;M198	8(M203)	+1450F	105.0	20/20	105,128	, 299	ı
(1) 8 INCH, M2A1	-	-350F	25.0	20/20	24.870	.055	0
(2) 8 INCH, M2A1	1	-350F	25.0	20/20	24,880	,057	0
(1) 8 INCH, M2A1	1/2	+700F	15,0	13/15	14.947	990	0
(2) 8 INCH, M2A1	1/2	+700F	15.0	15/15	14,943	.059	0
(1) 4.2 INCH, M30	7 INCR	+700F	13,0	15/15	13,064	.071	1.
(2) 4,2 INCH,M30	7 INCR	+700F	13.0	15/15	13.029	.059	ı
(3) 10°MM, M103	7	+700F	50.0	13/16	50.054	.073	0
(1) 155MM, M1	-	+700F	PD/FUNCT	20/20	ı	1	t
(2) 155MM,M1	7	+700F	PD/FUNCT	20/20	ı	i	1
(4) 155FM, M198	_	+700F	15.0	15/15	ı	1	ı

*LEGEND: (1) HAMILION LOT 83D000E108 - CONTROL RDS.

- HAMILTON LOT 83D000E107 RDS. ASSEMBLED W/PIN ASSEMBLED IN TIMER AND TRIGGER (ZIG-ZAG). (5)
- HAMILTON LOT 83D000E107 SEQ. ROUGH HANDLED 8RD5a-350F, 8 RDS a+1450F. 3
- HAMILTON LOT 83D000E113 (15) TEST RDS 5 EA RECOVERY VEHICLES WERE FIRED. (†)

13,14 MAY 1983

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EILMED A. C.

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